

INVESTIGATION OF BIT HYDRAULICS FOR  
GASIFIED DRILLING FLUIDS

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Approval of the Graduate School of Natural and Applied Sciences

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## **ABSTRACT**

### INVESTIGATION OF BIT HYDRAULICS FOR GASIFIED DRILLING FLUIDS

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Accurate determination of the pressure losses at the bit is very important for drilling practices in petroleum industry. In the literature, there are several studies on determination of the pressure losses. Major focus is concentrated on single phase drilling fluids, which is far from accurate estimation of pressure losses for multiphase fluids, i.e., fluids including a liquid and a gas phase, at the bit. Some of

these models are valid for multiphase fluids, however, they are either valid for very high gas flow rates, or developed using very strong assumptions.

This study presents a mathematical model for calculating bit hydraulics for gasified drilling fluids. The theory, which is valid for both sonic (critical) and subsonic (sub-critical) regimes, is based on the solution of the general energy equation for compressible fluid flow. The model is sensitive to changes in internal energy, temperature and compressibility. In addition, the model uses “mixture sound velocity” approach.

A computer program is developed based on the proposed mathematical model. The program calculates pressure drop through a nozzle in subsonic flow region, and suggest flow rate if the calculated pressure drop values is in the sonic flow pressure ranges.

The program has been run at reasonable field data. The results of the models have been compared with the results of existing models in the literature. The results show that the pressure losses through the bit can be estimated with a variation less than 9%. Also, it has been observed that bottom hole pressure, velocity of the liquid phase and nozzle size have a strong influence on bit pressure drop.

**Keywords:** Gasified Drilling Fluids, Bit Hydraulics, Sonic (Critical) and Subsonic (Sub-critical) Regimes, Multiphase Flow, and Sound Wave Transmission.

## ÖZ

### GAZ KARIŞIMLI SONDAJ SIVILARI İÇİN MATKAP HİDROLİĞİNİN İNCELENMESİ

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Petrol sektöründe yapılan sondaj operasyonlarında, matkaplardaki basınç kayıplarının kesin olarak belirlenmesi çok önemlidir. Literatürde, basınç kayıplarının hesaplanması üzerine yapılmış çalışmalar mevcuttur. Bunlardan birçoğu, tek fazlı sondaj akışkanları için kullanılan bağıntı ve yöntemler içermekte olup, çok fazlı sondaj akışkanları, bir başka deyişle sıvı gaz karışımları için, basınç düşümlerinin başarılı bir şekilde hesaplanmasını sağlamaktan çok uzaktır. Varolan

alıřmalardan birkaçı ise, ya yksek gaz akıř hızlarında uygun sonular ya da veren birok varsayıma dayanarak meydana getirilen alıřmalardır.

Bu alıřma, gazlı sondaj akıřkanları hidroliğini hesaplayacak bir matematiksel model sunmaktadır. Kritik ve kritik altı akıř rejimlerine uygun olan bu model, sıkıřtırılabilir akıř iin olan temel enerji denkleminde elde edilmiřtir. Model, i enerji, sıkıřtırılabilme arpanı ve sıcaklık deęiřimlerini dikkate almaktadır. Buna ek olarak, “karıřımların ses hızları” yaklařımı bu modelde kullanılmıřtır.

Geliřtirilen matematiksel model kullanılarak bir bilgisayar programı oluřturulmuřtur. Bu program, matkap basıncı kayıplarını ses altı blgesinde hesaplamakta ve eęer akıř ses st blgede ise ses, altı blgesine uygun akıř hızları nermektedir.

Bilgisayar programı, saha deęerlerine uygun verilerle alıřtırılmıřtır. Elde edilen sonular, literatrde olan modellerden elde edilen sonularla karřılařtırılmıř ve geliřtirilen modelin sonularının %9'dan az bir fark gsterdięi grlmřtir. Ayrıca, kuyu dibi basıncının, sıvı faz hızının ve nozul apının matkap basıncı kaybı deęerleri zerinde ok etkili oldukları gzlenmiřtir.

**Anahtar Kelimeler:** Gaz Karıřımlı Sondaj Sıvıları, Matkap Hidrolięi, Ses Altı (Kritik Altı) ve Ses st (Kritik) Rejimleri, ok Fazlı Akıř, Ses Dalgası Aktarımı.

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## NOMENCLATURE

$a$	Acceleration, ft/sec <sup>2</sup>
$A$	Area, ft
$C_p$	Isobaric heat capacity, (ft-lbf)/(lbm-°F)
$C_v$	Isotropic heat capacity, (ft-lbf)/(lbm-°F)
$C$	Medium compressibility, psi <sup>-1</sup>
$C_N$	Nozzle discharge coefficient, dimensionless
$d_c$	Nozzle diameter, ft
$d_{c-eq}$	Equivalent diameter of nozzles, ft
$d_d$	Pipe diameter downstream of the nozzle, ft
$D$	Dilation, dimensionless
$E$	Internal energy of one pound fluid, ft-lbf/lbm
$f$	Weight fraction in the flowing fluid, dimensionless
$F$	Heat capacity ratio, dimensionless
$g$	Acceleration of gravity, ft/sec <sup>2</sup>
$g_c$	32.2 (lbm-ft)/(lbf-sec <sup>2</sup> )
$k$	Heat capacity ratio
$ke$	Kinetic energy of one pound fluid, (ft-lbf)/lbm
$M$	Molecular weight, lbm/mol
$m$	Mass, lbm

$\dot{m}_c$	Calculated mass flow rate, lbm/sec
$\dot{m}_a$	Actual mass flow rate, lbm/sec
$NN$	Number of nozzles, dimensionless
$n$	Polytropic expansion exponent, dimensionless
$P$	Pressure, psi
$\bar{P}$	Average pressure, psi
$P_B$	Bit Pressure Drop across the bit , psi
$P_r$	Reduced pressure, dimensionless
$P_R$	Pressure drop ratio, dimensionless
$pe$	Potential energy of one pound fluid, (ft-lbf)/lbm
$PE$	Potential energy, ft-lbf
$Q$	Heat transferred to one pound of the flowing fluid, (ft-lbm)/lbm
$q$	Flow rate, cuft/sec
$q_{gsc}$	Flow rate at standard conditions, scft/sec
$R$	Universal gas constant (psi-ft <sup>3</sup> )/(lbm-mol-°R)
$s$	Condensation
$T$	Temperature, °F
$\bar{T}$	Average temperature, °F
$T_r$	Reduced temperature
$W$	Work, ft-lbf
$w$	Work of one pound fluid, (ft-lbf)/lbm
$V$	Volume, cuft
$z$	Gas compressibility factor, dimensionless

$Z$  Elevation, ft

### **Greek Letters**

$v$  Velocity, ft/sec

$\kappa$  Elasticity, psi

$\beta$  Volumetric fraction, dimensionless

$v$  Specific volume, ft<sup>3</sup>/lbm

$\rho$  Density, lbm/ft<sup>3</sup>

### **Subscripts**

$f$  fluid

$g$  gas

$l$  liquid

$m$  mixture

$BH$  bottom hole

0 initial

1 upstream of the nozzle

2 at the nozzle throat

3 condition just downstream of the nozzle throat if flow is subsonic

4 recovered condition downstream of polytropic compression.

## **CHAPTER 1**

### **INTRODUCTION**

Involved with highly fractured low pressure formations and depleted reservoirs, low pressure drilling, having several distinct advantages over conventional drilling, eliminates problems such as partial or total loss of circulation, formation damage, differential sticking, etc. In addition, it is proven that the low weighted drilling fluids increase the penetration rate and extend bit life.

Gasified drilling fluids have been increasingly used for drilling depleted reservoirs and highly fractured formations. Injecting gas and liquid through the drillstring generates the mixture drilling fluid. Gasification of the fluid through the drillstring leads to compressible two-phase flow in the drillstring, at the bit and at the annulus. It must be pointed out that, the effectiveness of bit could be improved by increasing in hydraulic power. The penetration rate would increase with increasing hydraulic horsepower until the cuttings were removed as fast as they were generated. In fact, the hydraulic power depends on pressure drop across the bit. Therefore, major focus has to be on the pressure drop at the bit.

The compressible two-phase fluid has a different flow behavior than widespread single phase drilling liquids. Flow patterns, compressibility factors and temperature dependence properties differentiate the two-phase fluids from single phase incompressible fluids. There have been many attempts to define gasified drilling fluid hydraulics. Flow of two phase fluids has been expressed using either empirical correlations or using mechanistic modeling approached. Like the proposed model in this study, almost the entire mathematical models presented in the literature are based on the general energy equation. Yet, some of those models have been developed for wellhead chokes. These models have been used to guide throughout this study. In deed, in the literature, developed mathematical models exist for estimating pressure drop at the drilling bits and those are simple forms of general energy equation. The proposed model has been compared with those mathematical models.

## **CHAPTER 2**

### **LITERATURE REVIEW**

Two-phase flow through a restriction may be either sonic (critical) or subsonic (sub-critical). Most of the models were developed for wellhead chokes and take sonic flow into consideration. A group of those models are comprised of empirical relationships (Omana [1], Ros [2, 3], Pilehvari [4], Osman and Dokla [5, 6]). These investigators used field data to propose a relationship to predict flow rate in the sonic region. The relationships basically consist of a three-parameter equation in which the flow rate is linearly proportional to the upstream pressure of the choke. These empirical models generally are valid over a range where experimental data were available, but they give poor results when extrapolated out of range.

The other models are theoretical relationships that have been derived from the basic fluid flow principles and they are given below.

As early as 1949, Tangren, et. al. [7] conducted the first significant study on two-phase flow through a restriction. They derived expressions for velocity of sound, equation of motion for two-phase flow and sonic flow through a choke. The

procedure, used in their analysis, was based on the basic laws of continuity, momentum, energy and ideal gas equation of state applied for a mixture.

Many years later, Ashford [8], and Ashford and Fierce [9], using a similar approach with Tangren et. al. [7], derived the general energy equation to predict the sonic limit pressure. The model of Ashford and Fierce [9] assumed that the derivative of mass flow rate with respect to the pressure ratio is zero at sonic flow. However, the model has an uncertainty and deficiency, those are, downstream pressure cannot be easily determined and the model fails in subsonic flow.

Sachdeva, et. al. [10] extended the analysis of Ashford and Fierce [9] and proposed an equation to predict the sonic pressure ratio through a choke. It was assumed that the gas phase at the entrance of the restriction is contracted isentropically, but at downstream of the restriction, the gaseous fraction of the fluid expands polytropically. Therefore, their equations of pressure drop need only one input pressure (upstream or downstream pressure).

Fortunati [11] used a new approach, and developed correlations for sound velocity, sonic and subsonic flow, and sonic and subsonic transition. He has introduced a sound velocity equation. The developed sound velocity equation, yet, cannot be applied for low gas/liquid ratio. Gould [12] pointed out that, for mass fraction of gas less than 0.4 are not applicable for Fortunati's work.

Perkins [13] and Clark and Perkins [14] reviewed the thermodynamic basis, and developed a theoretical framework, which is valid for both subsonic and sonic flow through a choke. In these studies, the framework constructed based on the general energy equation and sonic flow boundary could also be calculated in addition to sonic flow rate. Since the majority of these works are focused on correlating the available flow values at subsonic flow, their model is used as the major guide of the proposed model in this study. The model of Perkins' [13, 14] can determine variables at any point in the flowing system based on the following assumptions:

- Temperature varies with position,
- Velocity varies with position,
- The gas compressibility factor is constant,
- The liquids have a negligible compressibility compared to gas phase,
- Elevation changes are negligible,
- The flow process is adiabatic and frictionless,
- Internal energy changes are negligible

According to Perkins' [13, 14] study, the compressible two-phase fluid runs into a sudden expansion just at the throat of the choke. They developed an approximated relation, which depends on downstream pressure, and choke and downstream medium size. The relation is explained in detail in Chapter 4.

There are major differences between the proposed model in this study and Perkins' work [13, 14], which are factor of internal energy factor and changes in

gas compressibility factor. The proposed model is derived from the general energy equation including the effect of internal energy and compressibility factor changes. Moreover, in Perkins' work [13, 14], the reversible adiabatic expansion of an ideal gas (polytropic expansion) is inserted into those of compressible two-phase fluid. In the proposed model, in this study the polytropic expansion is applied to only gas phase.

Guo, Harelend and Rajtar [15] used a thermodynamics based approach, and explained the pressure drop equations for whole drill string and bit. In their work, aerated mud is used. As the other models, their model was also based on generalized energy equation. Moreover, they made common assumptions, such as elimination of energy losses due to friction and that of elevation changes. However, according to Lyons, Guo and Seidel [16], the application of this model is limited with high gas volume fractions. The developed equation of Guo et. al. [15] is presented in Appendix B.

Gücüyener [17], and Liu and Medley [18] obtained an implicit equation for the pressure drop through bit nozzles. The equation is applicable for gal/liquid mixtures. The equation was based on common equation, i.e., the general energy equation. In their studies, the upstream velocity of the fluid was assumed to be zero. Actually, the difference between the velocities of upstream and nozzle was significantly large. Grounded on that fact, the assumption is logical for low gas flow rates or small nozzle sizes. In addition to this assumption, their equation

ignores the internal energy term in the general energy equation. The development of their equation is presented in Appendix B.

## **CHAPTER 3**

### **STATEMENT OF THE PROBLEM**

In drilling operations, when gasified drilling fluids are used very little is known about the, estimation of the pressure drop through the bit nozzles.

In the literature, there are a few researches focusing on this problem. Most of the existing studies have developed models for wellhead chokes; indeed, many of them are not suitable for describing flow behavior of gasified drilling fluids through nozzles. Moreover, most of these models are valid for only sonic flow, where in drilling practices, the flow is subsonic. The applicable works for pressure drop and the bit are based on many strong assumptions. Although these approximations facilitate the development and usage of the developed equations, the results of those equations are not giving exact values.

This study is aimed to improve a model suitable for the flow behavior at the bit for gasified fluids which has been derived from the general energy equation by using fewer approximations. A mathematical model is developed for determining pressure drop at the bit, and estimating the appropriate flow rates for both liquid and gas phase to keep the mixture velocity at sub-sonic conditions. Basic

conservation equations, i.e., momentum and energy, are used. Several other hydraulics calculations, which are used to optimise the drilling performance, are conducted, such as, hydraulic horsepower, impact force and jet velocity. The calculated pressure data are compared with other models' results available in the literature. Also, a sensitivity analysis is conducted to understand the impact of several drilling parameters on pressure drop at the bit.

## **CHAPTER 4**

### **THEORY**

A bit nozzle can be treated as a restriction in a pipe “space”. There may exist both sonic and subsonic flow in the restriction. During sonic flow, the flow rate through the restriction reaches a maximum value with respect to the prevailing upstream conditions. The velocity of the fluids flowing through the restriction reaches sonic (pressure wave propagation) velocity [10]. Indeed, the value of sonic velocity depends on the fluid properties. In the case of sonic flow, large turbulent energy losses are ultimately possible because of shock front [13]. This shock front is the same as the sonic wall. As a result, even if the downstream pressure is decreased, the flow rate does not increase or increase in the upstream. Pressure does not affect the flow rate, so does the downstream pressure, see Figure 1. If the upstream pressure is decreased, the flow rate does not change until the sonic – subsonic boundary is reached. If the decrease in the upstream pressure continues and the pressure is lower than the sonic pressure limit, the flow rate will start to be affected.

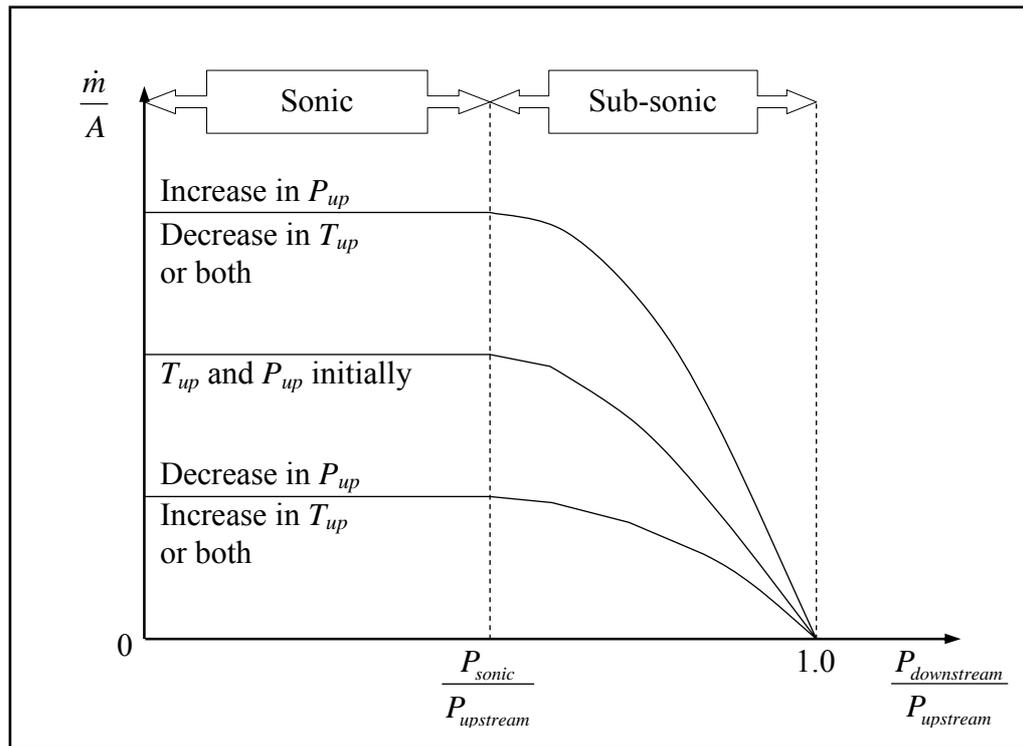


Figure 1 Pressures vs. Flow Rate Relations At a Nozzle

According to Perkins [13], the general energy equation is valid for both sonic and subsonic flow. For subsonic flow, the general energy equation shows that a polytropic expansion (reversible adiabatic expansion) occurs as the fluid accelerates downstream of the restriction. Sachdeva et. al. [10] stated that during gas expansion at the throat, a temperature gradient develops between the phases, resulting in fast heat transfer between them. This process is in-between the extremes of isothermal and adiabatic processes. Thus, a polytropic process approximates the heat flow in the gas-liquid mixtures.

In an adiabatic process, there is no heat transfer with the surroundings. Therefore, for an adiabatic process, the energy takes the form as

$$dU = -PdV \dots\dots\dots(4.1)$$

If the system is closed and the volume is kept constant, the energy equation can also be expressed as,

$$dU = C_v dT \dots\dots\dots(4.2)$$

Since the system is adiabatic, substituting eq. (4.1) into (4.2) gives

$$- PdV = C_v dT \dots\dots\dots(4.3)$$

If the gas phase of the fluid is assumed to be ideal gas, eq. (4.3) takes the form

$$PV = RT = (C_p - C_v)dT \Rightarrow C_v dT = (C_v + C_p) \frac{T}{V} dV \dots\dots\dots(4.4)$$

Separateing the variables and writing  $k$  representing  $C_p/C_v$  gives

$$\frac{dT}{T} + (k - 1) \frac{dV}{V} = 0 \dots\dots\dots(4.5)$$

Integration of eq. (4.5) yields

$$TV^{k-1} = \text{constant} \dots\dots\dots(4.6)$$

Equation (4.6) implies how temperature and volume of an ideal gas vary during an adiabatic expansion. If the gas expands, the temperature goes up. Indeed, the pressure varies also, and the ideal gas conforms to the relation given below

$$\frac{PV}{T} = \text{constant} \dots\dots\dots (4.7)$$

Eliminating the temperature term and expressing eq. (4.6) and on the basis of one pound of flowing gas yields

$$PV_g^k = \text{constant} \dots\dots\dots (4.8)$$

As mentioned before, polytropic expansion occurs at downstream conditions. Therefore, the variables in eq. (4.8) are properties of gas inside the nozzle. Since the system has two phases, which include liquid and gas, the heat capacity ratio of the fluid is

$$n = \frac{(f_g C_{pg} + f_l C_{pl})}{(f_g C_{vg} + f_l C_{vl})} \dots\dots\dots (4.9)$$

So, for two-phase fluids eq. (4.8) takes the form

$$PV_g^n = \text{constant} \dots\dots\dots (4.10)$$

Perkins [13] defined the sonic – subsonic flow. To determine the sonic boundary, it is required to know the magnitude of the discharge pressure in the restriction throat,  $P_2$ , as presented in Figure 2. The pressure at this point is not normally measured directly. On the other hand, pressure at the outlet of the nozzle throat,  $P_3$ , can be estimated by a pressure measurement device. For subsonic flow, Perkins [13] has developed an approximate relationship,

$$P_3 = P_1 - \frac{P_1 - P_4}{1 - (d_c/d_d)^{1.85}} \dots\dots\dots(4.11)$$

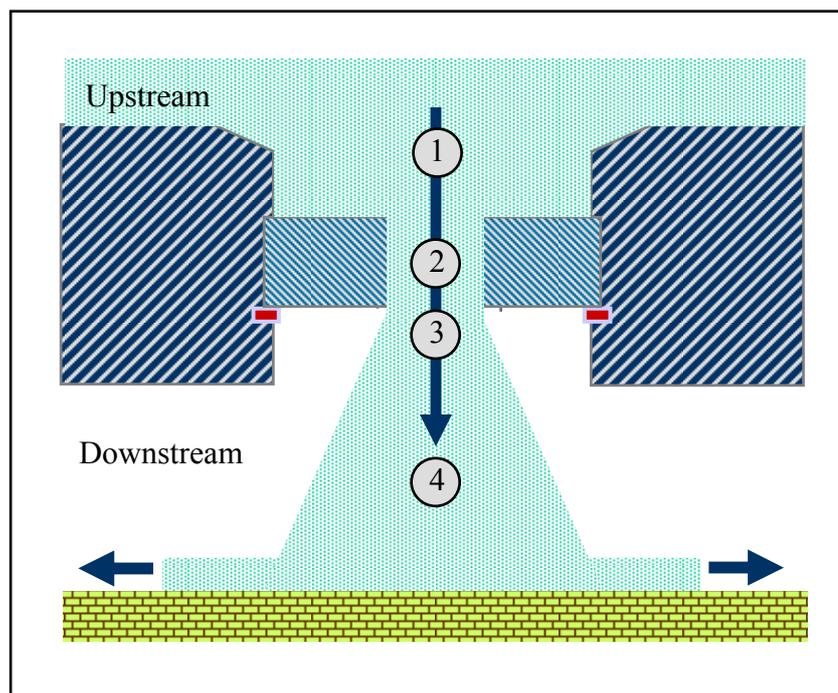


Figure 2 Diagram of Flow Through a Nozzle

#### 4.1. The General Energy Equation For Sonic And Sub-Sonic Flow

The thermodynamic framework for multiphase sonic flow is based on the principles of conservation of energy. As a gasified fluid mixture approaches to a restriction, its velocity will increase due to the reduction in the pressure. For any point in the flowing system, the following assumptions are made:

1. Velocity varies with axial (flow) direction, but for a particular arbitrary point, all components are moving with the same velocity. According to Fortunati [11], when dealing with the two-phase flow flow, all the researchers found that both phases will have the same velocity, if these conditions are satisfied;
  - Velocity is greater than 32.8 ft/sec
  - Froude Number,  $Fr > 600$ , where

$$Fr = \left( \frac{v^2}{g_c d} \right) \dots\dots\dots(4.12)$$

Moreover, Ros [2, 3] showed that there is practically no slippage at between the phase the throat. Hence, it is reasonable to assume same velocity for each phase at the nozzle.

2. The gas compressibility factor is constant.

3. The liquid has a negligible compressibility when compared with gas.
4. Elevation changes are ignored.
5. The flow is adiabatic. Even the nozzle is not insulated; the process is so fast that the rate of heat transfer between the fluid flowing in the nozzle and the surroundings is zero. Besides, the fluid does not have enough time for any significant heat transfer inside the nozzle.
6. The flow is frictionless. According to Ros [2, 3], the wall shear forces can be neglected. Moreover, in the Omana's [1] experimental work, viscosity has a negligible effect on the pressure drop.
7. Temperature varies in the axial direction, but for an arbitrary point, all components are moving with the same temperature.

#### **4.2. Introduction to The General Energy Equation**

One of the fundamental laws of nature is the conservation of energy principle. It states that during an interaction, energy can change from one form to another, but the total amount of energy remains constant. That is, energy can neither be created nor destroyed; it can only change form. The first law of thermodynamics is simply an expression of the conservation of energy principle.



$$w_{flow} = P v = W_{flow} / m \dots\dots\dots(4.15)$$

In field units, eq. (4.15) can be expressed as

$$w_{flow} = P v \text{ [psia] [ft}^3\text{/lbm]} \Rightarrow \text{[lbf/in}^2\text{] [ft}^3\text{/lbm]} \Rightarrow 144 \text{ [lbf/ft}^2\text{] [ft}^3\text{/lbm]} \dots\dots(4.16)$$

which reduces to

$$w_{flow} = 144 P v \text{ [ft-lbf/lbm]} \dots\dots\dots(4.17)$$

#### 4.2.2. Internal Energy

It is related to the molecular structure of a system and the degree of the molecular activity, and they are independent of outside reference boundary on the control volume in Figure 2. The individual molecules in a system move with same velocity, vibrate, and rotate about an axis during their random motion. The internal energy of a system is determined by

$$E = C_v T \text{ [ft-lbf/lbm-}^\circ\text{F] [}^\circ\text{F]} \Rightarrow \text{[ft-lbf/lbm]} \dots\dots\dots(4.18)$$

### 4.2.3. Kinetic Energy

Basically, it is the energy that the system possesses as a result of its motion relative to a reference boundary on the control volume, as seen in Figure 2. From the point of view of macroscopic scale, the kinetic energy of flowing fluids is significant. On the basis of one pound of flowing fluid, this can be written as

$$ke = \frac{0.5m\vartheta^2}{g_c} [(ft/sec)^2] [1/(lbm-ft/lbf-sec^2)] \Rightarrow [ft-lbf/lbm] \dots\dots\dots (4.19)$$

### 4.2.4. Potential Energy

It is the energy that the system possesses as a result of its elevation in a gravitational field. In this study, potential energy is ignored. When compared with the other energy components, it is too small to take into consideration.

$$PE = mgZ \dots\dots\dots (4.20)$$

On the basis of unit mass of flowing fluid, eq. (4.20) can be written as

$$pe = \frac{gZ}{g_c} [ft/sec^2] [ft] [1/(lbm-ft/lbf-sec^2)] \Rightarrow [ft-lbf/lbm] \dots\dots\dots (4.21)$$

#### 4.2.5. Heat Transfer

Heat is defined as the form of energy that is transferred between two systems by virtue of a temperature. It only takes place, if there is a temperature difference. As stated before, in this study, the process is so fast that the fluid does not have sufficient time for any heat transfer, thus, it is negligible.

$$Q = C_v \Delta T \dots\dots\dots(4.22)$$

#### 4.2.6. Work

Work is the energy transfer associated with force acting through a distance. The energy can cross the boundary of the closed system only in the form of heat or work. Actually, a rising piston, a rotating shaft, and an electrical wire cross having current on the system boundaries are all associated with work interactions. In the system, in which the drilling fluid flowing through a nozzle, there is no work done by the system. Therefore, in the proposed model, the work equals to zero.

### 4.3. Derivation of The General Energy Equation

The mathematical modelling of two – phase flow can be developed on the bases of the general energy equation. The system used, in this model, consists of a restriction. In the flow direction, there are four important points, as presented in Figure 2.

In Figure 2, the upstream of the nozzle is signed as point 1. At that point, velocity is low when compared with velocity at point 2. At point 1, although the pressure value is an unknown at this moment, kinetic and internal energies can be used to determine  $P_1$ . The fluid passes to point 2, i.e. flows through the nozzles. Since the fluid accelerates, there is kinetic energy increase through the nozzle. The energy changes form from flow work to kinetic energy, and the flow work term decreases. This means that, the fluid is decompressed and polytropic expansion occurs. Because the process is fast, there is no heat transfer to and from the surroundings. Therefore, at point 2, the temperature reduces due to the sudden adiabatic expansion. Unfortunately, pressure cannot be measured at point 2. Actually, pressure at the outlet of the nozzle throat,  $P_3$ , can be estimated by a pressure measurement device. During the flow of fluid, for subsonic flow, there is no change in pressure at this stage, i.e.,  $P_3$  equals to  $P_2$ .

The fluid, finally, reaches to point 4, where the fluid has the final state of energy, pressure and temperature values. Since a portion of kinetic energy is lost at

the discharge of the nozzle due to the geometrical expansion, the pressure difference at point 4 and point 3 is small. The relation between  $P_3$  and  $P_4$  is expressed by eq. (4.11)

The energy changes of the fluid are explained, before the fluid pass from point 1 to point 2. On the basis of one pound flowing fluid, the general energy equation can be written as

Initially, because of the small size of the nozzle, the difference in elevation is negligible, i.e.,  $Z_1 \cong Z_2$ . Then, since, there is neither external work done on the system nor internal work done by the system, work equals to zero. Moreover, the heat transfer is assumed to be negligible, thus  $W = 0$  and  $Q \cong 0$ .

The change in the internal energy can be expressed as

$$dE = d(C_v T) \dots\dots\dots(4.24)$$

where the heat capacity of the fluid constant volume is a function of temperature.

Thus, the general energy , eq. (4.23) equation reduces to

$$144(Pdv + vdP) + d(C_{vf} T) + \frac{1}{2g_c} 2v dv = 0 \dots\dots\dots(4.25)$$

For a control volume,  $dv$  in eq. (4.25) equals to 0. Then eq. (4.25) simplifies to

$$144(vdP) + d(C_{vf}T) + \frac{\vartheta d\vartheta}{g_c} = 0 \dots\dots\dots(4.26)$$

Integrating the equation (4.26), with respect to  $P, V$  and  $\vartheta$  yields

$$144 \int_{P_1}^{P_2} v dP + \int_{T_1}^{T_2} C_{vf} dT + \frac{1}{g_c} \int_{\vartheta_1}^{\vartheta_2} \vartheta d\vartheta = 0 \dots\dots\dots(4.27)$$

The first term of equation (4.27), specific volume of fluid,  $v$ , is a function of pressure. As mentioned above, the polytropic expansion occurs as the fluid accelerates through the nozzle. The polytropic expansion equation in terms of the specific volume of gas, confining pressure,  $P$ , polytropic expansion constant,  $b$ , and ratio of specific heat constant  $n$ , is given below.

$$P v_g^n = P(v_f - v_l)^n = b \dots\dots\dots(4.28)$$

where  $n$  is expressed as in eq. (4.9)

As the heat capacity of incompressible liquid at constant pressure is constant and equals to that at constant volume, eq. (4.9) takes the form

$$n = \frac{(f_g C_{pg} + f_l C_{vl})}{(f_g C_{vg} + f_l C_{vl})} \dots\dots\dots(4.29)$$

Specific volume of the fluid is defined as

$$v_f = (b/P)^{1/n} + v_l \dots\dots\dots(4.30)$$

The polytropic expansion equality for between point 1 and point 2 can be written as

$$P_1(v_{m1} - v_l)^n = P_2(v_{m2} - v_l)^n \dots\dots\dots(4.31)$$

So, the first term of eq. (4.27) becomes

$$\begin{aligned} 144 \int_{P_1}^{P_2} v_f dP &\Rightarrow 144 \int_{P_1}^{P_2} [(b^{1/n} P^{-1/n}) + v_l] dP \\ &\Rightarrow 144 b^{1/n} \left( \frac{n}{n-1} \right) \left( P_2^{\frac{n-1}{n}} - P_1^{\frac{n-1}{n}} \right) + 144 v_l (P_2 - P_1) \end{aligned} \dots\dots\dots(4.32)$$

The third term of eq. (4.25) is integrated.

$$\frac{1}{g_c} \int_{v_1}^{v_2} v dv = \frac{v_2^2 - v_1^2}{2g_c} \dots\dots\dots(4.33)$$

Integrating eq. (4.32) and eq. (4.33) into eq. (4.27) yields

$$144 b^{1/n} \left( \frac{n}{n-1} \right) \left( P_2^{\frac{n-1}{n}} - P_1^{\frac{n-1}{n}} \right) + 144 v_l (P_2 - P_1) + \left( \int_{T_1}^{T_2} C_{vf} dT \right) + \frac{v_2^2 - v_1^2}{2g_c} = 0 \dots\dots(4.34)$$

Combining eq. (4.30) and (4.34) gives

$$144(P_2(v_{f2} - v_l)^n)^{1/n} \left( \frac{n}{n-1} \right) \left( P_2^{\frac{n-1}{n}} - P_1^{\frac{n-1}{n}} \right) + 144v_l(P_2 - P_1) + \frac{v_2^2 - v_1^2}{2g_c} + \left( \int_{T_1}^{T_2} C_{vf} dT \right) = 0 \quad \dots\dots\dots(4.35)$$

After simplifying eq. (4.35) the final form of general energy equation becomes

$$144(v_{f2} - v_l) \left( \frac{n}{n-1} \right) \left( P_2^{\frac{n-1}{n}} P_1^{1/n} \right) + 144P_2(v_{f2} - v_l) \left( \frac{n}{n-1} \right) + 144v_l(P_2 - P_1) + \frac{v_2^2 - v_1^2}{2g_c} + \left( \int_{T_1}^{T_2} C_{vf} dT \right) = 0 \quad \dots\dots\dots(4.36)$$

In eq. (4.36), the mass flow rate term is isentropic and at all cases, the isentropic value is equal to the ratio of actual flow rate and nozzle discharge coefficient. According to Perkins [13], after a compilation of 1,432 data sets obtained from the literature, comprising both sonic and subsonic flows, the best overall average value of the discharge coefficient,  $C_N$ , is found to be 0.826.

#### 4.4. Velocity of Wave Transmission

The sound velocity is influenced by the physical properties of the medium, such as density and elasticity, which are equivalent to the mass as stiffness factor in the case of vibration of a particle, as stated by Wood [19]. While a wave of compression and rarefaction moves through a medium, the density (the volume)

fluctuates locally. The value of these fluctuations depends on the properties mentioned above as well as the applied force (power of sound). To clarify the behaviour of sound wave, the dilation, condensation, elasticity and density of the medium are examined.

- Dilation,  $D$ : It is the ratio of the increment of the volume to the original volume.

$$D = \frac{\Delta V}{V_0} \Rightarrow V = V_0(1 + D) \dots\dots\dots(4.37)$$

- Condensation,  $s$ : It is the ratio of the increment of density to the original density.

$$s = \frac{\Delta \rho}{\rho_0} \Rightarrow \rho = \rho_0(1 + s) \dots\dots\dots(4.38)$$

Conservation of mass is expressed as,  $\rho V = \rho_0 V_0 \Rightarrow (1 + s)(1 + D) = 1$ , so, condensation term,  $s$ , equals to  $-D$ . Here,  $sD$  is so small that the value is neglected.

- Volume elasticity,  $\kappa$ : The elasticity is the change in the volume of a compressible fluid, when the fluid is exposed to a pressure. The value of the initial volume is taken into consideration.

$$\kappa = -V_0 \frac{\partial P}{\partial V} \Rightarrow \kappa = -\frac{\partial P}{D} = \frac{\partial P}{s} \dots\dots\dots(4.39)$$

- Compressibility,  $c$ : The compressibility is the reciprocal of elasticity.

$$c = -\frac{1}{V_0} \frac{\partial V}{\partial P} \Rightarrow c = \frac{1}{\kappa} \dots\dots\dots(4.40)$$

Consider the case of plane waves travelling along the x-axis, as shown in Figure 4.

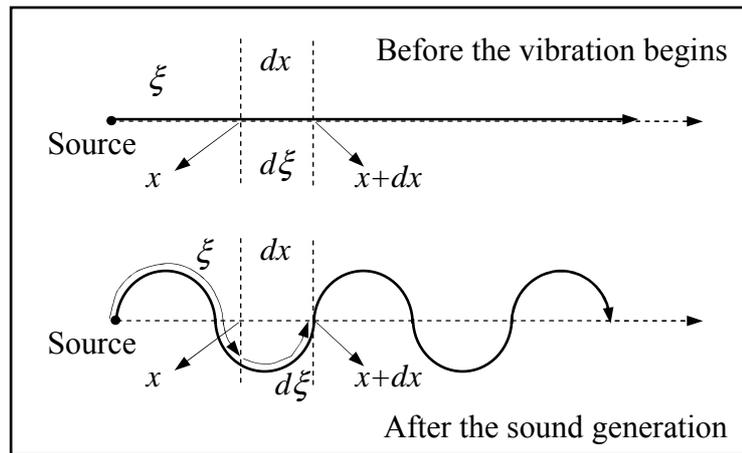


Figure 4 The Case of Plane Wave

Displacement of the planes normally at  $x$  and  $(x+dx)$  will be  $\xi$  and  $\left(\xi + \frac{d\xi}{dx} \partial x\right)$ .

When the source generates sound, the thickness of the imaginary layer changes between the plane  $x$  and plane  $x+dx$ . Before the generation of the round,

change of the thickness equals to  $dx$ , but then, the sound leads to an oscillation where the thickness is  $dx+d\xi$ . In other words, the difference is  $d\xi$ . So, the dilation becomes,

$$D = \frac{d\xi}{dx} = -s \dots\dots\dots(4.41)$$

If the excess pressure on the  $x+dx$  plane is  $dP$  (change in the force of the wave due to friction), there is a velocity decrease and the force is equal to

$$F = ma \Rightarrow dP = -(\rho_0 dx) \frac{d^2 \xi}{dt^2} \dots\dots\dots(4.42)$$

where  $\rho_0 \cdot dx$  is the mass of the unit area of the medium, and  $\frac{d^2 \xi}{dt^2}$  is the deceleration term.

Using eq. (4.40) and (4.42), one can derive

$$\frac{d^2 \xi}{dt^2} = -\rho_0 \frac{dP}{dx} \Rightarrow \frac{d^2 \xi}{dt^2} = \frac{\kappa}{\rho} \frac{d^2 \xi}{dx^2} \Rightarrow v_s = \sqrt{\frac{\kappa}{\rho_0}} \dots\dots\dots(4.43)$$

Eq. (4.43) can be applied for all cases of transmission of small amplitude plane wave in solid, liquid or gaseous media. In the gaseous - liquid medium, it can be assumed that  $P\rho_0 = \rho P_0 = \text{constant}$  for a constant temperature. So, the elasticity is reduced to

$$\kappa = -V_0 \frac{\partial P}{\partial V} \Rightarrow \kappa = \rho_0 \frac{\partial P}{\partial \rho} = \rho_0 \frac{P_0}{\rho_0} = P_0 \dots\dots\dots(4.44)$$

This, from eq. (4.43), the sound velocity becomes

$$v_s = \sqrt{\frac{P_0}{\rho_0}} \dots\dots\dots(4.45)$$

If the system is are not isothermal, but adiabatic, Using eq. (4.10), eq. (4.44) becomes

$$P V^n = P_0 V_0^n \Rightarrow P = \rho^n \frac{P_0}{\rho_0^n} \Rightarrow \frac{dP}{d\rho} = n\rho_0^{n-1} \left( \frac{P_0}{\rho_0^n} \right) \Rightarrow \kappa = nP_0 \dots\dots\dots(4.46)$$

Thus, from eq. (4.43), the sound velocity for compressible fluid mixtures can be written as

$$v_s = \sqrt{n \frac{P_0}{\rho_0}} \dots\dots\dots(4.47)$$

#### 4.5. Flow Diagram of The Model

1. The model requires the following inputs.
  - Bit inside and outside diameters used in equation (4.11),
  - Nozzle amount and nozzle size to calculate equivalent size of the nozzles,
  - Gas flow rate, in standard conditions, and liquid flow rate to estimate mass flow rates and mass fraction of the phases.
  - Bottom hole temperature and pressure.

The fluid inside drill string gains heat from the surroundings and when the fluid reaches to bit it is assumed that the temperature of the fluid is equal to the bottom hole temperature. Thus, bottom hole temperature is referred as  $T_1$ .

2. If the bit has more than one nozzle, the model estimates an equivalent nozzle diameter, as given in eq. (4.48).

$$d_{c-eg} = \sqrt{\sum_{i=1}^{NN} d_c^2} \dots\dots\dots(4.48)$$

3. As a first iteration, the value of downstream pressure,  $P_1$  is equalised to  $P_4$ .
4. Calculate  $P_3$ , by using eq (4.11) (Perry Relationship [14]) and at subsonic velocity. Calculated pressure value is assumed to be equal to nozzle pressure, i.e.,  $P_2=P_3$ ,

5. In order to determine the mass fraction, mass flow rates and specific volume of liquid and gas phases, physical properties of gas phase and liquid phase and mixtures of these phases should be well defined. The proposed model requires gas compressibility. The compressibility factor of air can be calculated by using eq. (A.1). Then, using  $z_1$  value obtained from eq. (A.1),  $v_{f1}$  and  $\rho_{g1}$  can be calculated as

$$P_1 V_{g1} = n_{g1} R T_1 z_1 \Rightarrow \rho_{g1} = \frac{P_1 M_g}{R T_1 z_1} \dots\dots\dots (4.49)$$

and

$$\rho_f = \frac{m_f}{V_1} = \frac{m_f}{m_f \left( \frac{f_g}{\rho_g} + \frac{f_l}{\rho_l} \right)} = \frac{1}{\left( \frac{f_g}{\rho_g} + \frac{f_l}{\rho_l} \right)} = \frac{1}{v_f} \dots\dots\dots (4.50)$$

respectively.

6. Calculate the polytropic expansion exponent,  $n$ , by using eq. (4.29). In this equation the mass fractions and constant volume heat capacities of gas and liquid fraction of the fluid are desired. The model needs air and water heat capacity values. The specific heat capacity values for air are listed in Appendix A. The estimated values of air are used for correlating heat capacity at desired

$P$  and  $T$ . For water, Perkins [13] suggested 778 ft-lbf/lbm-°F or constant volume heat capacity.

7. Calculate  $T_2$  at the nozzle using eq. (4.51). As mentioned in earlier, a polytropic expansion (reversible adiabatic expansion) occurs as the fluid accelerates downstream of the nozzle. Expansion causes the temperature through the nozzles to drop. Equation (4.6) implies how temperature and volume of an ideal gas vary during an adiabatic expansion. For fluid flow across a bit, equation (4.6) takes for, as seen in equation (4.51). Using the equation (4.51), the temperature in the nozzle can be calculated as

$$(T_2 + 460) = (T_1 + 460)P_R^{(n-1)/n} \dots\dots\dots(4.51)$$

8. Calculate  $z_2$ ,  $v_{f2}$  and  $\rho_{g2}$  as mentioned in step 5.
9. Calculate the estimated mass flow rate. Due to the conservation of mass, at all points in and out of the bit, the mass flow rate should be constant. Mass flow rate can be calculated using

$$\dot{m}_c = \dot{m}_a / C_N \dots\dots\dots(4.52)$$

where, the actual mass flow rate is proportional to volumetric fluid flow rate and it calculated by eq. (4.53),

$$m_a = (q_l \times \rho_l) + (q_g \times \rho_g) \dots\dots\dots(4.53)$$

Thus, velocity of the fluid at any point of the bit could be calculated as shown in eq. (4.54)

$$v = \frac{m_c}{A} \left( \frac{f_g}{\rho_g} + \frac{f_l}{\rho_l} \right) \dots\dots\dots(4.54)$$

10. Determine the average pressure and temperature values, and recalculate the polytropic expansion exponent,  $n$ , as mentioned in step 6. In the final form of general energy equation, eq. (4.36), polytropic expansion exponent is important. Although it varies with the temperature and the pressure of the fluid, it is assumed as constant for flow through the bit. Therefore, an average value of the exponent is required and is estimated by using average temperature and pressure values, such that

$$\bar{P} = 0.5(P_1 + P_2) \dots\dots\dots(4.55)$$

and

$$\bar{T} = 0.5(T_1 + T_2) \dots\dots\dots(4.56)$$

respectively.

11. Use the general energy equation eq. (4.36) for calculating  $P_1$ .

12. Go back to step 4 and iterate the  $P_1$  values until the results converge. If the percent difference between the values of  $P_1$  from values is less than  $1.0 \times 10^{-6}$ , the model gives the best  $P_1$  values. The error analysis presents in Appendix C.

13. Check  $v_2$  value, whether the system is in subsonic flow range or not. If  $v_{sonic} > v_2$  than the flow is subsonic.

$$v_{sonic} = \sqrt{n \frac{P_2}{\rho_{f2}}} \dots\dots\dots(4.57)$$

14. If the velocity is greater than sonic velocity, the program reduces the gas and liquid flow rates at the same proportion till the velocity go below the sonic velocity.

## **CHAPTER 5**

### **RESULTS AND DISCUSSIONS**

In this study, a mathematical model for estimating the pressure drop of the bit for two-phase fluids is presented. In the literature, there are some similar applications, however the diversity of the proposed model is to be more sensitive to flow conditions and inner forces. The algorithm and related calculated process gives more accurate results than the models previously developed.

The developed model could work in the subsonic – sonic flow conditions. Since sonic flow is not convenient with drilling operations, the model obtains pressure drop results across the bit only in subsonic flow. If the flow is sonic, the proposed model suggests a change in liquid or gas flow rates, so that the flow is kept in subsonic flow ranges.

For making calculations of the pressure drop faster, a computer program is developed based on the proposed model. Using the developed program, runs are performed at widely encountered field conditions for various air-water mixture flow rates. These values were used for comparison of the performance of the

proposed model with the previously developed models and determination of the influences of flow parameters on pressure drop. The conditions are listed in Table 2, where Case 1 is assigned as a base condition. The conditions were selected to explain the effects of nozzle size, diameter of the bit, liquid and gas flow rates, bottom hole pressure and bottom hole temperatures. The influences of these factors have been discussed in the further section.

Table 1 Selected Parameters for Sample Run

	<b>Case 1</b>	<b>Case 2</b>	<b>Case 3</b>	<b>Case 4</b>
Gas Phase	Air	Air	Air	Air
Liquid Phase	Water	Water	Water	Water
$C_N$	0.826	0.826	0.826	0.826
Num. Of Nozzle	3	3	3	3
Nozzle Size, in	13/32	13/32	15/32	13/32
Diameter of downstream, in	12 ¼	12 ¼	12 ¼	8 ½
Diameter of upstream, in	2 ¼	2 ¼	2 ¼	1 ¾
$Q_{GSC}$ , scft/min.	200, 500, 750, 1000 and 1500	200, 500, 750, 1000 and 1500	200, 500, 750, 1000 and 1500	200, 500, 750, 1000 and 1500
$Q_{liq}$ , gal/min	100, 200 and 300	100, 200 and 300	100, 200 and 300	100, 200 and 300
$P_{BH}$ , psia	250, 500, 750, 1250 and 1750	250, 500, 750, 1250 and 1750	250, 500, 750, 1250 and 1750	250, 500, 750, 1250 and 1750
$T_{BH}$ , °F	100 °F	150 °F	100 °F	100 °F

## 5.1. Flow Type Determination

As mentioned in theory, the proposed model initially assumes that, the flow pattern of two-phase fluids through the bit nozzles is dispersed bubbly. In the literature, some experiments were conducted to identify five distinct flow patterns for vertical upward flow: bubbly, dispersed bubble, slug, churn and annular flow, and two phase flow pattern type map for vertical flow were obtained, as presented Figure 5. The flow pattern of a gasified fluid can be estimated by calculating the superficial liquid and gas velocities. For this study, the calculated superficial values of liquids and gas phase flowing through bit nozzle were checked on that map, and it was observed that, for all cases, the flow is in the disperse flow area.

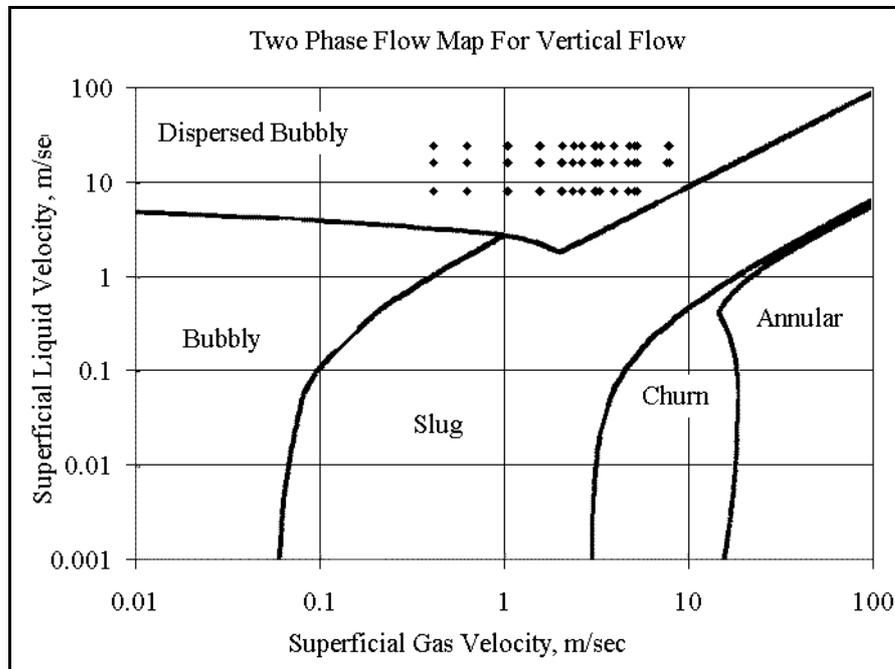


Figure 5 Two Phase Flow Map For Vertical Flow, Kaya et. al. [20]

The results satisfy the assumption, the flow of two-phase fluid across the bit nozzles is disperse bubbly flow. In Figure 5, the points in the dispersed flow area show the results obtained in this study.

As mentioned before, it is assumed that, velocity varies with axial (flow) direction, but for a particular arbitrary point, all components are moving with the same velocity. Moreover, it is also assumed that, no slippage occurs between the phases at the nozzle throat. Two boundaries are specified to confirm these assumptions. These boundaries are Froude limit, eq. (4.12), and 32.8 ft/sec velocity of fluid limit at the nozzle throat ([11] and [2, 3]). For this study, the calculated velocity through the bit nozzle and Froude number were checked, as seen in Figure 6, and it is observed that calculated values are greater than these limits. Thus, for all cases, there is no slippage between the phases at the throat, i.e., air and water fractions are moving with same velocity.

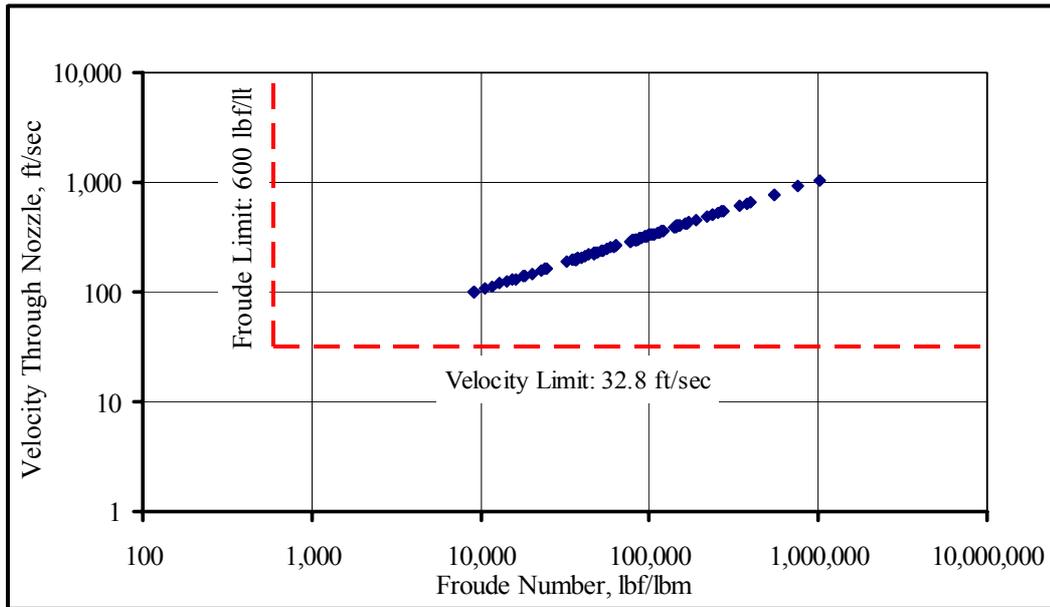


Figure 6 Froude Number and Fluid Velocity Through Nozzle

## 5.2. Sensitivity Analysis of Hydraulic Parameters

Effects of bottom hole pressure and temperature, liquid flow rate, gas flow rate, and nozzle and bit sizes on bit pressure drop were investigated. The effective two-phase drilling fluid hydraulic parameters, i.e., nozzles size, flow rates, etc., could change the efficiency of the bit and lead to obtain more efficient bit horsepower.

### 5.2.1. Effect of Bottom Hole Pressure on Bit Pressure Drop

The bottom hole pressure has a significant effect on pressure drop at the bit. As the pressure difference between upstream and down stream of the nozzles

increases, the fluid tends to move faster while passing through the nozzle, yet, the nozzle does not let high velocity through itself. Then, inside the bit, the fluid accumulates, which leads an increase in the upstream pressure. The proposed model values, which were obtained for different bottom hole pressures, were plotted on pressure drop vs. gas flow rate (Figure 7) and pressure drop vs. bottom hole pressure graphs (Figure 8). As seen from Figure 7, as the bottom hole pressure decreases, the pressure drop values increase. The bottom hole pressure could rise until the upstream pressure of the nozzle reaches to sonic pressure boundary.

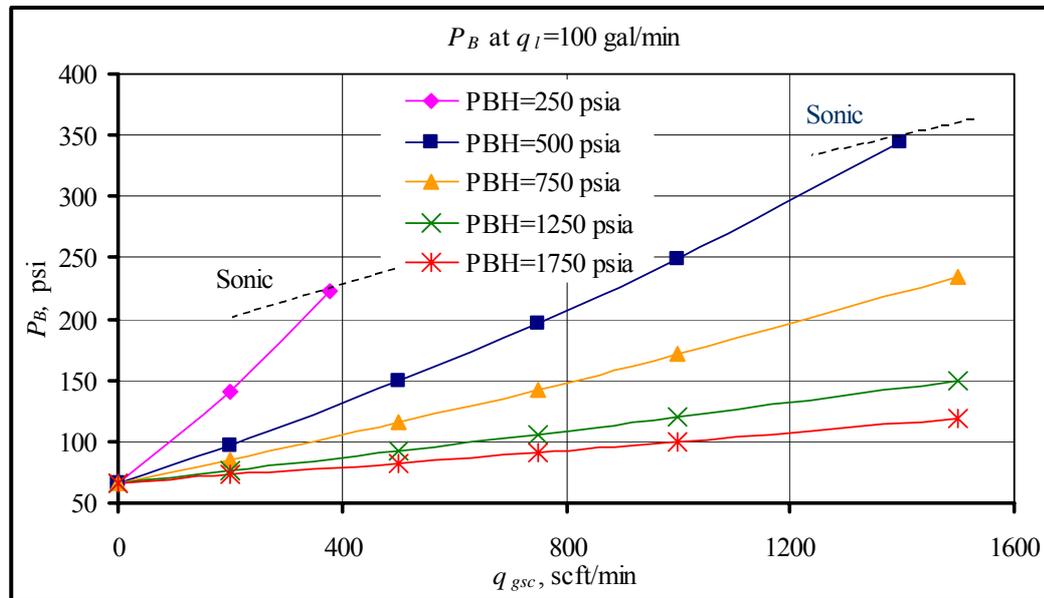


Figure 7  $P_{BH}$  Effect on  $P_B$  at  $q_l=100$  gal/min

At higher-pressure values, the flow is sonic. At that region, the flow pressure behaviour depends on the compressibility of the fluid. The detailed information is given in section 5.2.6.

The effect of bottom hole pressure on pressure drop at the bit is significant as seen in Figure 8. At 200 scft/min gas flow rate, the bit pressure drop curve shows a steep increase, as the bottom hole pressure is started to reduce from 500 psia to 250 psia. At higher gas flow rate values, the effect of the  $P_{BH}$  is more significant. As seen in Figure 8, 1000 scft/min curve shows more severe decrease in  $P_B$  values with increasing bottom hole pressure. The difference between the curves increase as the gas flow rates increase. As a result, at higher gas flow rates, the bottom hole pressure has more considerable effect on  $P_B$ .

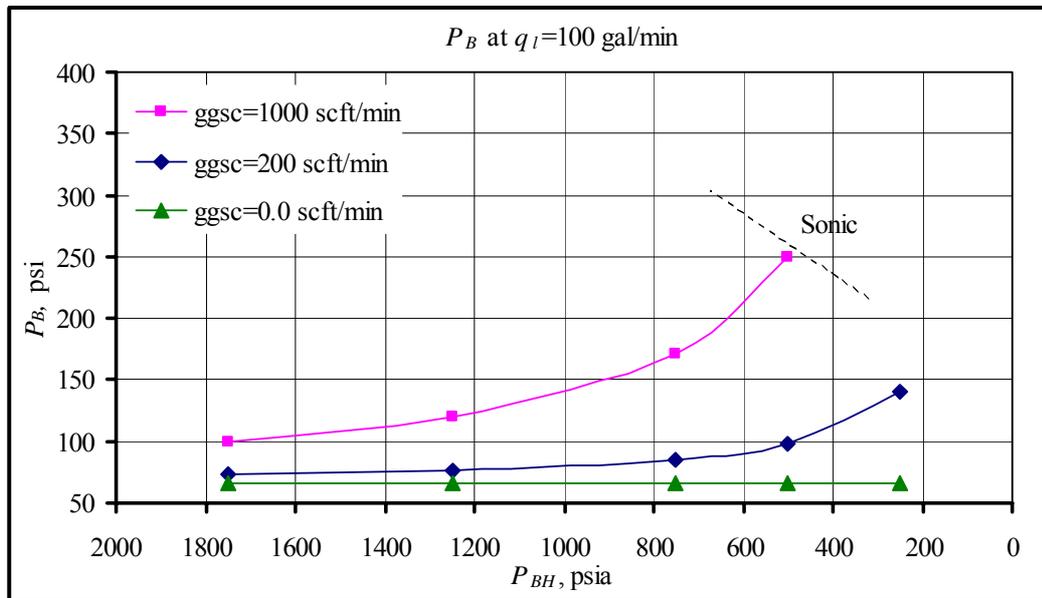


Figure 8  $P_{BH}$  Effect on  $P_B$  at  $q_l=100$  gal/min and Constant  $q_{gsc}$

In Figure 8, when there is no gas flow (0.0 scft/min), the curve is flat. Since only the gas phase in the system is assumed to be compressible, the bottom hole pressure is effective only on the compressible phase. Thus, as gas flow rate approaches to 0.0 scft/min, this effect disappears. Therefore, an increase in the gas

concentration causes  $P_{BH}$  to be more effective on  $P_B$ . If the fluid is incompressible,  $P_{BH}$  has no effect on  $P_B$ .

### 5.2.2. Effect of Flow Rate on Bit Pressure Drop

The flow rates have a noticeable effect on bit pressure drop. For constant gas flow rates and bottom hole pressures, as the liquid rate is increased, a significant increase in bit pressure drop is observed, as seen in Figure 9.

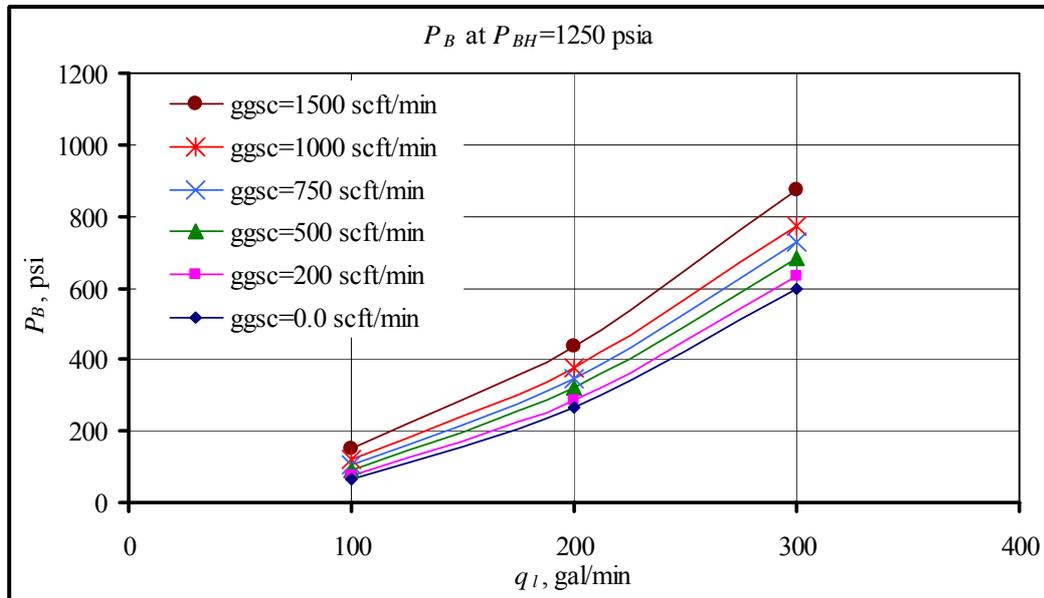


Figure 9 Liquid Flow Rate Effect on  $P_B$  at  $P_{BH}=1250$ psia and Constant  $q_l$

The influence of gas flow rate on  $P_B$  also is observed in Figure 10. Actually, change in the liquid flow rate has stronger influence on bit pressure drop than the change in gas flow rate. The compressibility of the gas leads to a reduction in the volume. Therefore, even the gas flow rate in the system is doubled, that does not

result to a significant increase in the bit pressure drop. On the other hand, as seen in Figure 9, when the liquid flow rate is doubled, the bit pressure drop values are triplet.

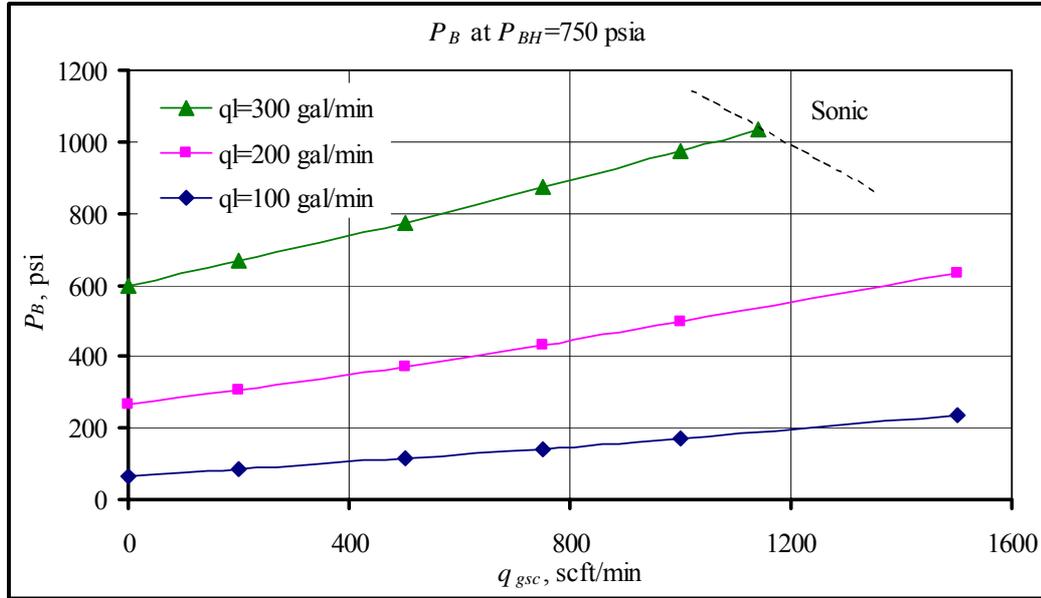


Figure 10 Gas Flow Rate Effect on  $P_B$  at  $P_{BH}=750$ psia and Constant  $q_{gsc}$

In Figure 10, at higher values of gas flow rate, the pressure in the bit reaches the sonic limit. Since the total mass flow rate increases, inside the nozzle, the velocity of the fluid reaches the sonic velocity of that fluid. Indeed, since the sonic velocity limit is higher in liquid than in gas, the liquid concentration of the two-phase fluid rises, so the sonic velocity of the mixtures and the sonic pressure limit. However, if the total flow rate is high enough, the flow will reach to its sonic limit, even the sonic velocity of the fluid is increased. Therefore, mixture flow rate could be increased easily till the sonic pressure limit is reached

### 5.2.3. Effect of Bottom Hole Temperature on Bit Pressure Drop

Temperature is a function of gas properties. Changes in temperature influence only the gas phase, provided that incompressible liquid assumption is defined. The density, compressibility and heat capacities of the gas depend on temperature. However, temperature effect is not as significant as the effect of bottom hole pressure or flow rates on bit pressure drop. The calculated values of the proposed model were plotted on bit pressure drop vs. gas flow rate (Figure 11) and bit pressure drop vs. liquid flow rate (Figure 12) graphs for a constant bottom hole pressure.

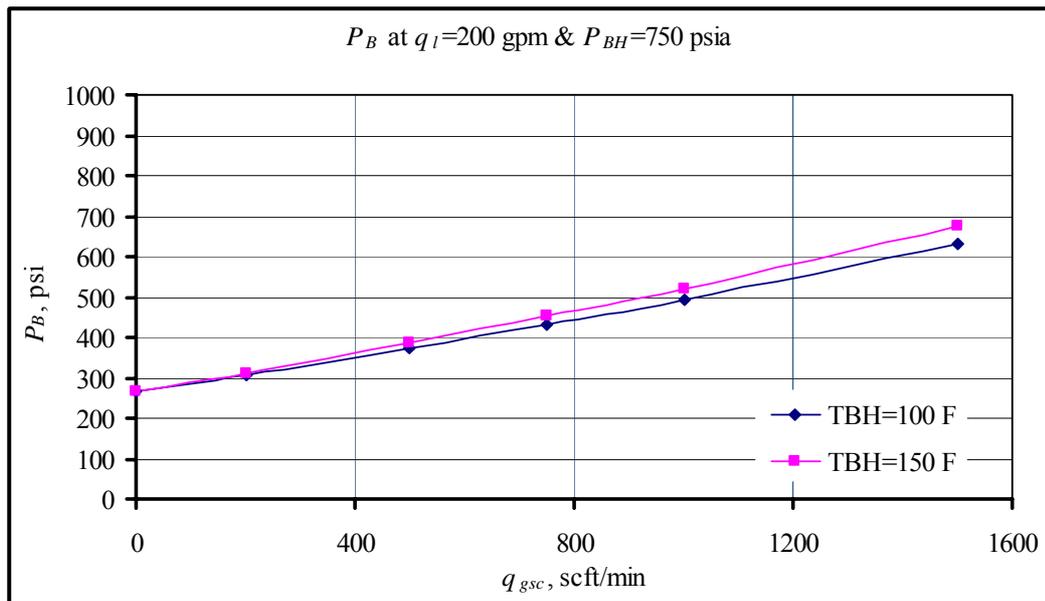


Figure 11 Temperature Effect on  $P_B$  at  $q_l=200$ gpm and  $P_{BH}=750$ psia

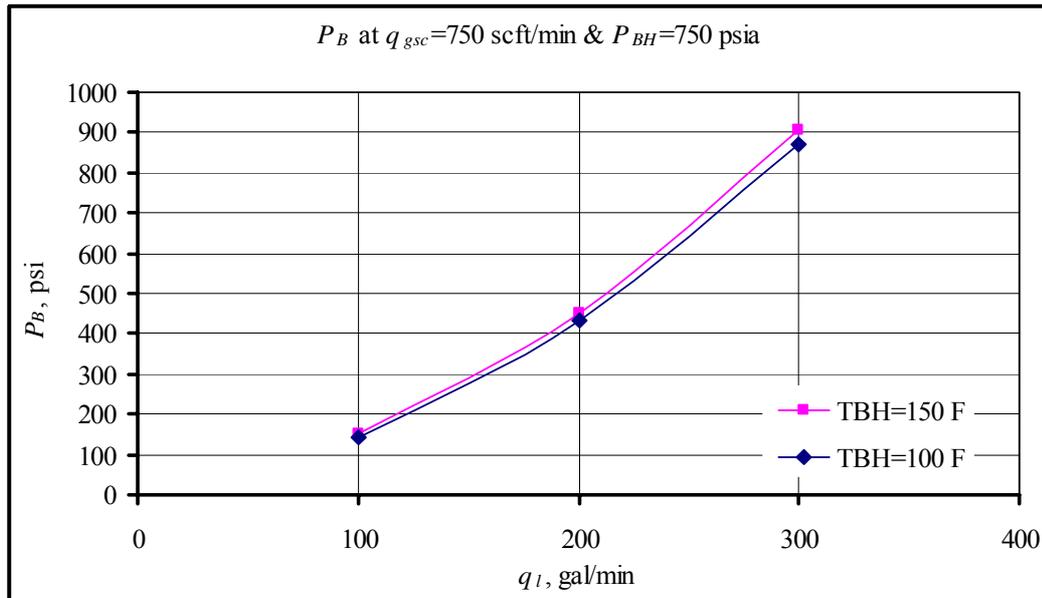


Figure 12 Temperature Effect on  $P_B$  at  $q_{gsc}=750$ scft/min and  $P_{BH}=750$ psia

In Figure 11, at lower gas flow rates, i.e. low gas concentration, the curves are collapsing on each other. In other words, when the gas flow rate increases, the gas concentration rises in the mixture for a constant liquid flow rate, and the temperature changes affect  $P_B$  more due to the increase the amount of gas.

As seen in Figure 12, temperature changes result in a slight change in pressure drop values. Increase in liquid flow rate does have a significant effect on bit pressure drop. As mentioned above, temperature is a function of gas properties, and changes in temperature do not affect the bit pressure drop for different liquid flow rate values.

#### 5.2.4. Effect of Nozzle Size on Bit Pressure Drop

Nozzle size leads to a considerable change in bit pressure drop. In fact, in theory, the nozzle size influences the velocity through the nozzle. For a smaller nozzle size, the fluid cannot pass easily through the nozzle. For this reason, the fluids begin to accumulate in the bit and cause an increase in the upstream pressure even smaller value of flow rates, as presented in Figure 13 and Figure 14. As seen in Figure 13, for a constant liquid flow rate and bottom hole pressure, as the nozzle size is increased, a significant decrease in pressure drop is observed for all gas flow rates. Moreover, as presented in Figure 14, when the gas flow rate is kept constant, bit pressure drop is even sharper for smaller nozzle size as the liquid rate is increased.

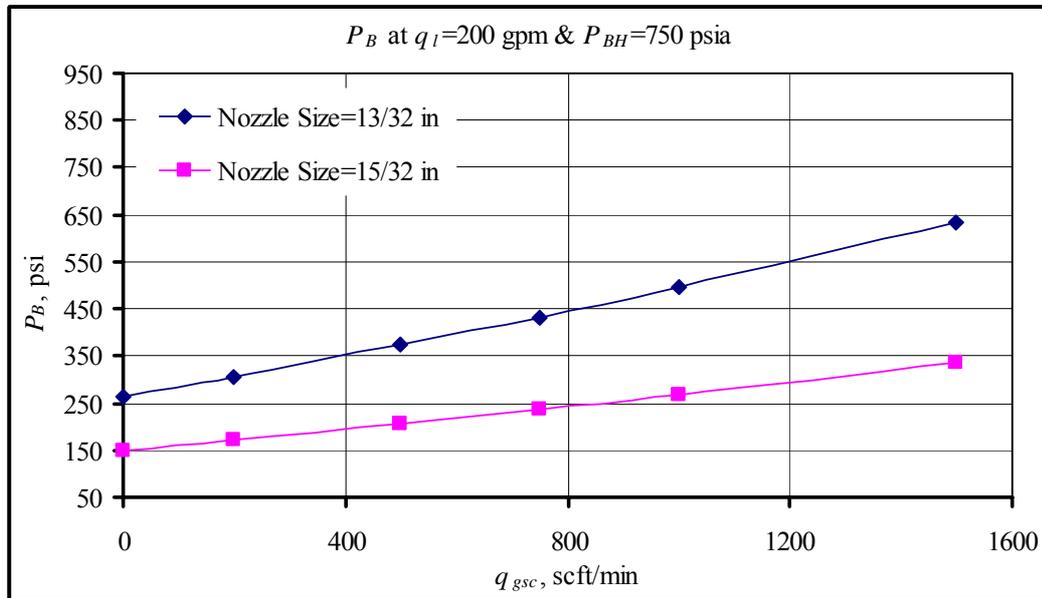


Figure 13 Nozzle Size Effect on  $P_B$  at  $q_l=200$  gpm and  $P_{BH}=750$ psia

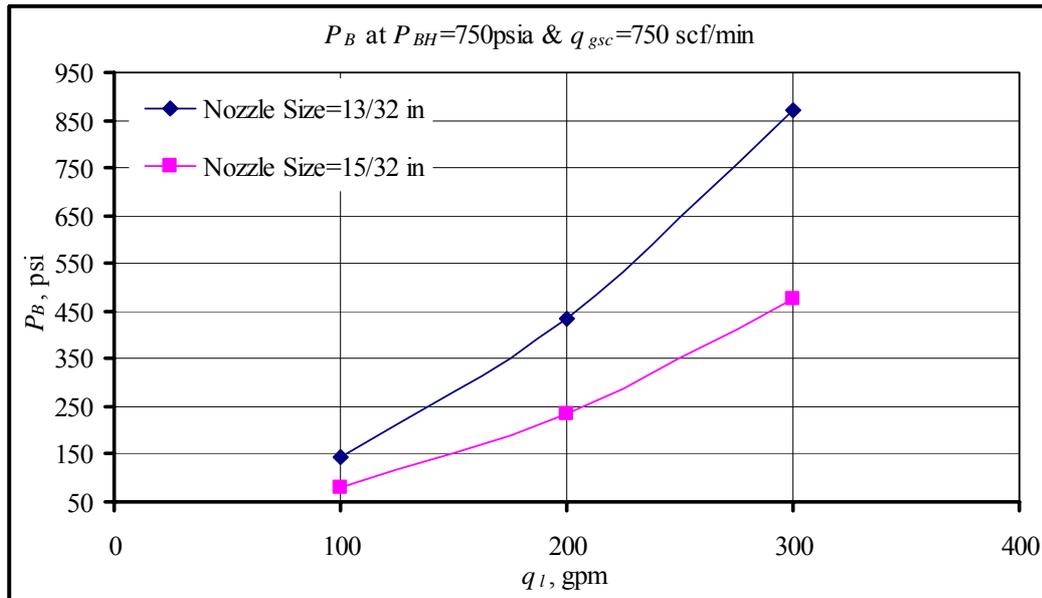


Figure 14 Nozzle Size Effect on  $P_B$  at  $q_{gsc}=750$  scft/min and  $P_{BH}=750$ psia

### 5.2.5. Effect of Bit Size on Bit Pressure Drop

Bit size does not have any influence on the bit pressure drop values. Actually, the bit size value has to be taken into consideration during the estimation of annular hydraulic calculations, which is out of the scope of this work. Difference of the proposed model results for different bit sizes, is insignificant. From Figure 15, it can be seen that there is no change in the bit pressure drop when the bit size changes. As a result, it can be said that, the bit size does not have any effect on bit pressure drop.

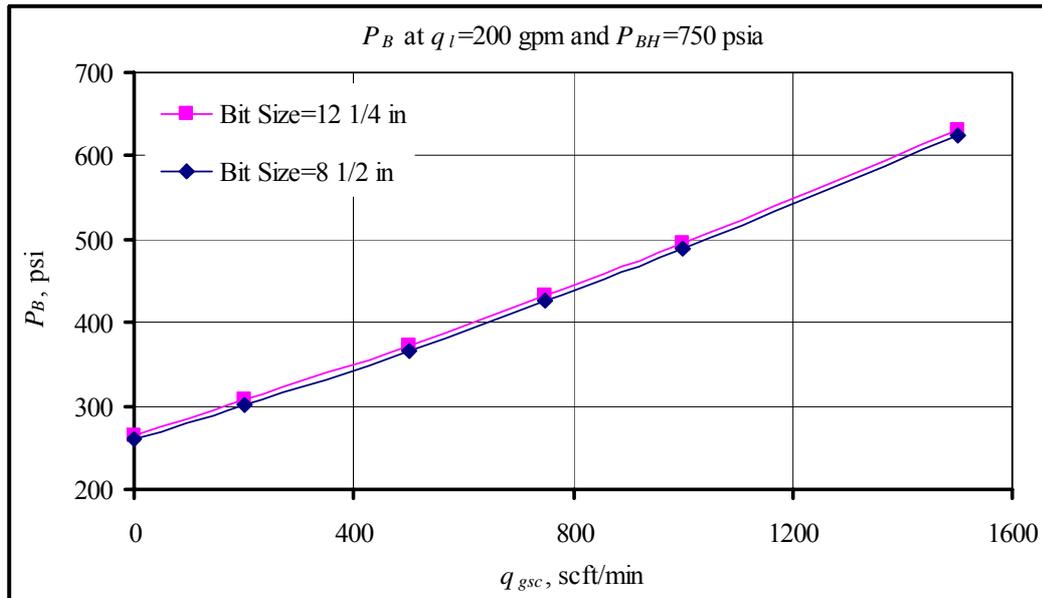


Figure 15 Bit Size Effect on  $P_B$  at  $q_l=200$  gal/min and  $P_{BH}=750$ psia

### 5.2.6. Effect of Sonic Pressure on Bit Pressure Drop

The fluid flowing through a nozzle cannot have a higher velocity than the sonic velocity of that fluid. When the fluid velocity through the nozzle reaches to the sonic velocity, accumulation of the fluid in the bit increases. At such high velocities, there exists a sonic wall, which prevents the fluid to flow with very high velocities. This results a build up in the upstream pressure. In other words, the bit pressure values rise. For a two-phase fluid system, the pressure drop values increase rapidly as the flow rates are approaching to the sonic boundary. As presented in Figure 16, as the gas flow rate increases, the velocity in the nozzle rises and this leads an increase in upstream pressure. Since the bottom hole

pressure is kept constant, the upstream pressure continues to increase with increasing gas flow rate.

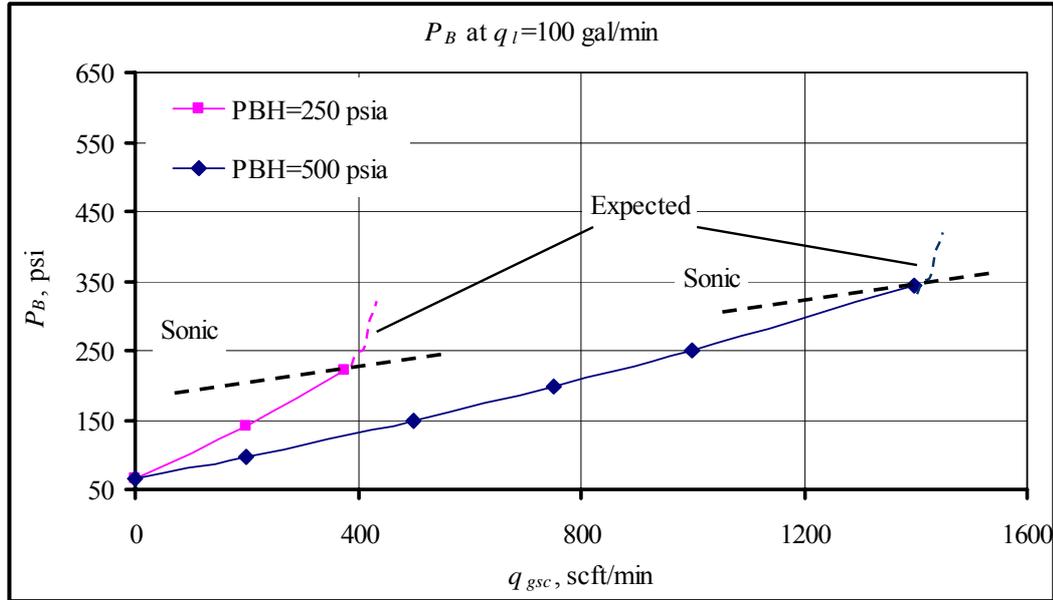


Figure 16 Sonic Pressure Effect on  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=750$ psia

When the fluid velocity reaches to the sonic velocity, there is an excessive accumulation inside the bit. The accumulation leads to the upstream pressure to increase rapidly; so the gas is compressed and the velocity will decrease. The accumulation and compression changes are instant and vanish with the stabilization of the upstream pressure. In Figure 16, expected bit pressure drop curve for a two-phase fluid at sonic conditions was plotted. For an incompressible fluid flowing through a bit nozzle, as the flow velocity approaches to the sonic velocity, due to the incompressibility of the fluid, a sudden increase in the upstream pressure is observed.

At lower bottom hole pressure values, the fluid can reach the sonic limit even at lower gas flow rates. If the sonic limits, in Figure 16, are compared, there is more than a 100 psi difference for lower and higher gas flow rates. At lower bottom hole pressure, the fluid has a higher volume than at high pressure due to compressibility of the gas phase. So, the volumetric flow rate of the fluid increases. Moreover, at low pressures, the sonic velocity of the fluid is low. As a result, at different bottom hole pressures, the reasons why the sonic pressure varies are the sonic velocity changes and compressibility of the gas phase.

### **5.3. Comparison of The Proposed Model With Previously Developed Models**

Bit pressure drop predictions of the proposed model are compared with the existing ones, which are Gücüyener [17] and Liu [18] (eq. (A.4)) and Guo [15] (eq. (A.5)). This comparison was conducted in a range of practical values of bottom hole pressures, gas and liquid flow rates, as listed in Table 1. The comparison indicates that the differences between the models are very sensitive to changes in bottom hole pressure, gas and liquid flow rates.

Internal energy and temperature changes of flow of two-phase fluid, which flows across a bit, are not included in the previous models. The proposed model does not ignore those changes for the bit pressure drop calculations. The existence of those factors affects the pressure drop values. The variations due to those factors are also examined.

Deviation of the existing models from the exact solutions obtained in this study is analyzed and variation analysis results are presented in tables 4 to 16 and figures 24 – 36.

### **5.3.1. Effect of Bottom Hole Pressure on Bit Pressure**

The comparison of the models in the literature and proposed model was conducted for a constant liquid rate (100 gal/min) and varying bottom hole pressure and gas flow rates. The results are plotted on graphs, as seen from Figure 17 to Figure 21.

In these figures, the proposed model does not present significant difference with Gücüyener [17] and Liu [18] curves. However, significant differences are observed with the results of Guo [15] Model. The results of the comparisons are presented in Appendix E. The proposed model has a maximum of 9% difference from the Gücüyener [17] and Liu [18] model results.

The effect of internal energy on bit pressure drop is not significant for lower bottom hole pressure, but higher bottom hole pressure values, variation becomes more significant. In Figure 17, the proposed model curves, one which does not include the internal energy, is different from the proposed model with the internal energy. The proposed model without internal energy is estimating closer results to Gücüyener [17] and Liu [18] model curves. Since, their model does not include the internal energy and it assumes that the velocity of the fluid at the bit,  $v_1$ , is zero,

the predicted results of the proposed model without internal energy and Gücüyener [17] and Liu [18] model show a similar behavior. In the general energy equation, eq. (4.36), the kinetic energy change depends on the inlet and outlet velocities of the fluid at the bit. Due to Gücüyener [17] and Liu [18] assumption of zero inlet velocity of fluid at point 1, the kinetic energy values at upstream condition are zero and calculated kinetic energy gain is higher for their model. This leads to a higher bit pressure drop value. In Figure 17, if Gücüyener [17] and Liu [18] model curve is compared with the proposed model, which does not take the internal energy into consideration, are listed and analyzed in Appendix E. According to the variation analysis, the existence of internal energy and factor of temperature changes in the bit have a maximum 3.3% difference, as seen in Table 8 of Appendix E.

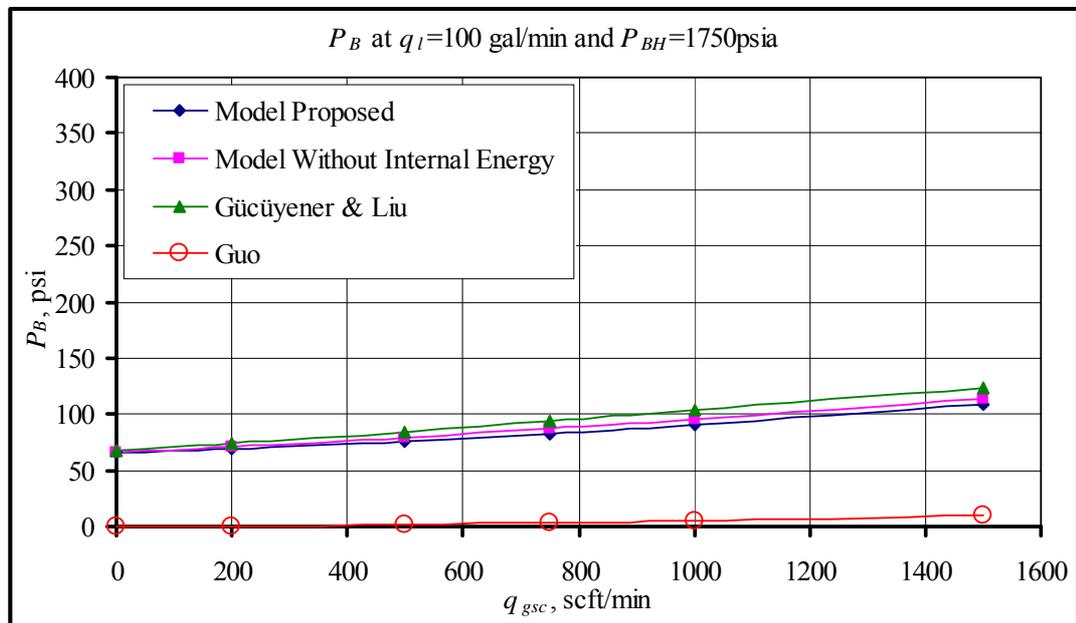


Figure 17  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=1750$ psia

The curves are collapsing on each other with increasing bottom hole pressure as seen from Figure 18 to Figure 21. However, the predicted values of Guo [15] model are approaching to zero with an increase in bottom hole pressure, as presented in Figure 17. Actually, as mentioned in literature survey chapter, the Guo [15] model is valid for high gas concentrations. Therefore, the pressure drop values rise at high gas flow rates, as seen Figure from 18 to 21. However, since the existing gas is compressed and the volume of the gas in the mixture is reduced, i.e., the gas concentration reduces, Guo [15] equation curves approach to zero.

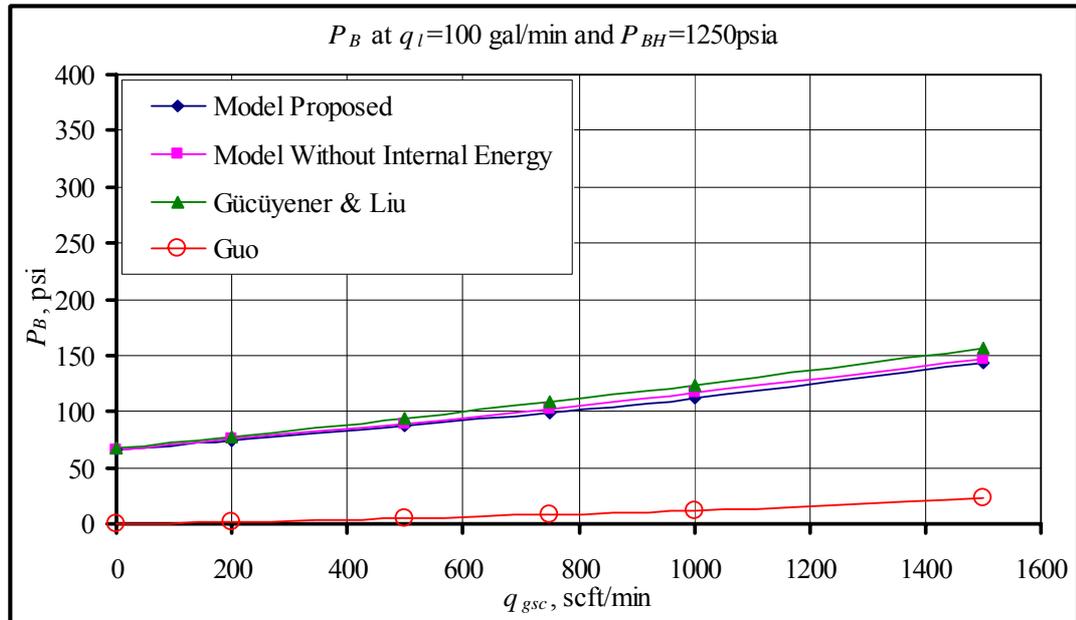


Figure 18  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=1250$ psia

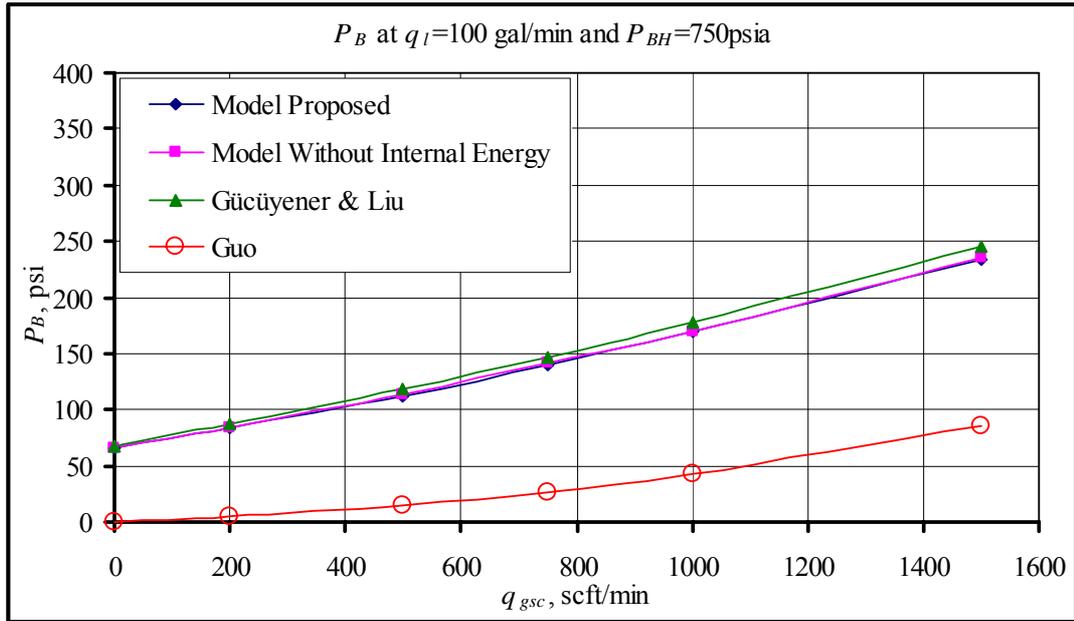


Figure 19  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=750$ psia

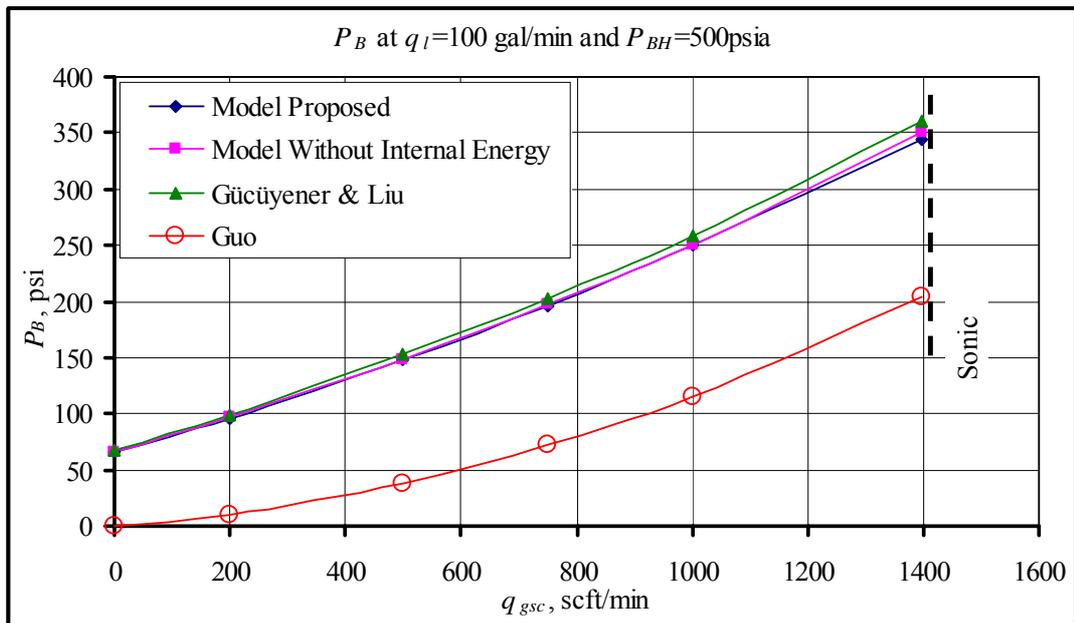


Figure 20  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=500$ psia

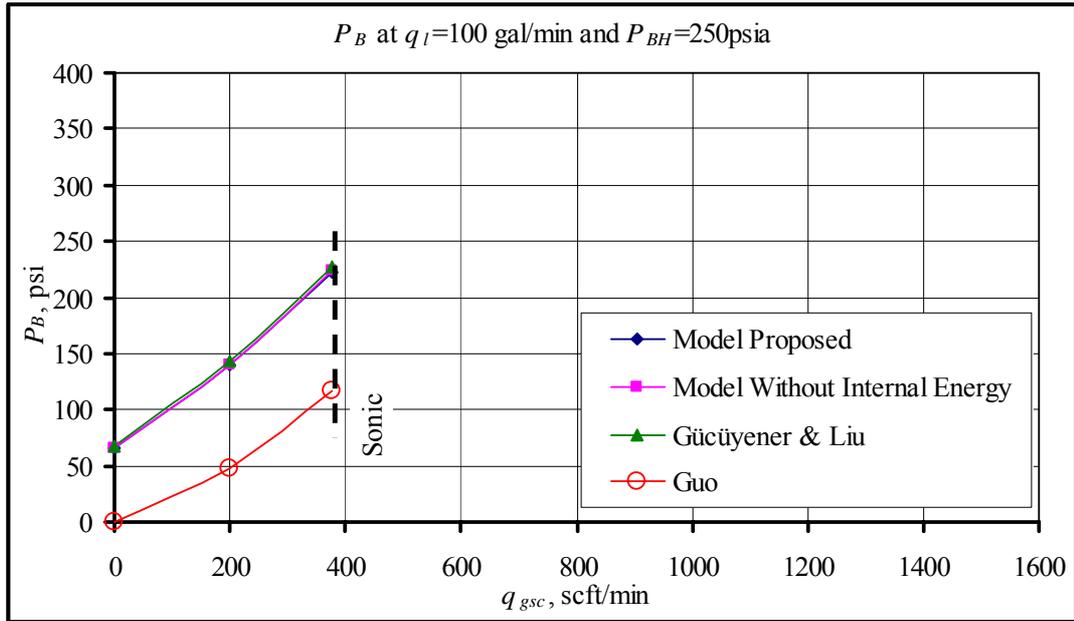


Figure 21  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=250$ psia

### 5.3.2. Effect of Liquid Flow Rate on Pressure Drop

The influence of the liquid flow rate on bit pressure drop has been discussed in the previous sections. In Figures 22 and 23, the effects of bit pressure drop varying liquid rates at constant bottom hole pressure on bit pressure drop are examined. The increase in the liquid flow rate results an increase in the bit pressure drop values, as presented in Figure 22 and Figure 23.

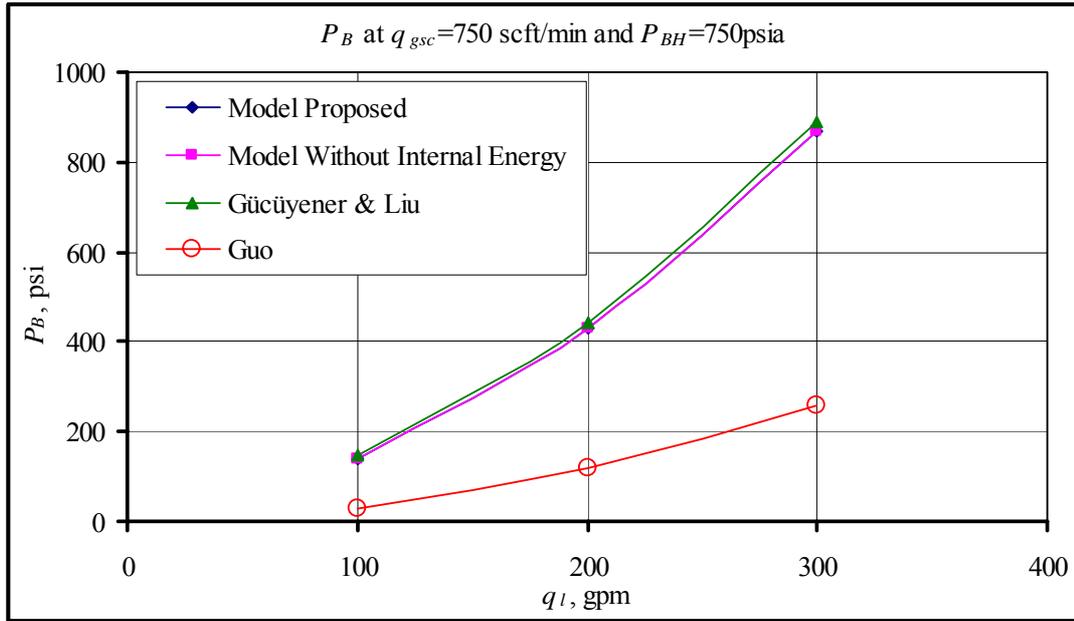


Figure 22  $P_B$  Values at  $q_{gsc}=750$  scft/min and  $P_{BH}=750$ psia

The curves in Figure 22 are approaching to each other except curve of Guo [15] model. The results of Gücüyener [17] and Liu [18] and the proposed model are close to each other. But, although the quantity is small, the variation increases with an increase in liquid flow rate. Moreover, the internal energy effect on pressure drop does not have a significant effect, as seen in Figure 22. But in Figure 23, the existence of internal energy in the proposed model has a small effect on pressure drop values. In fact, the bottom hole pressure is higher in Figure 23 than in Figure 22.

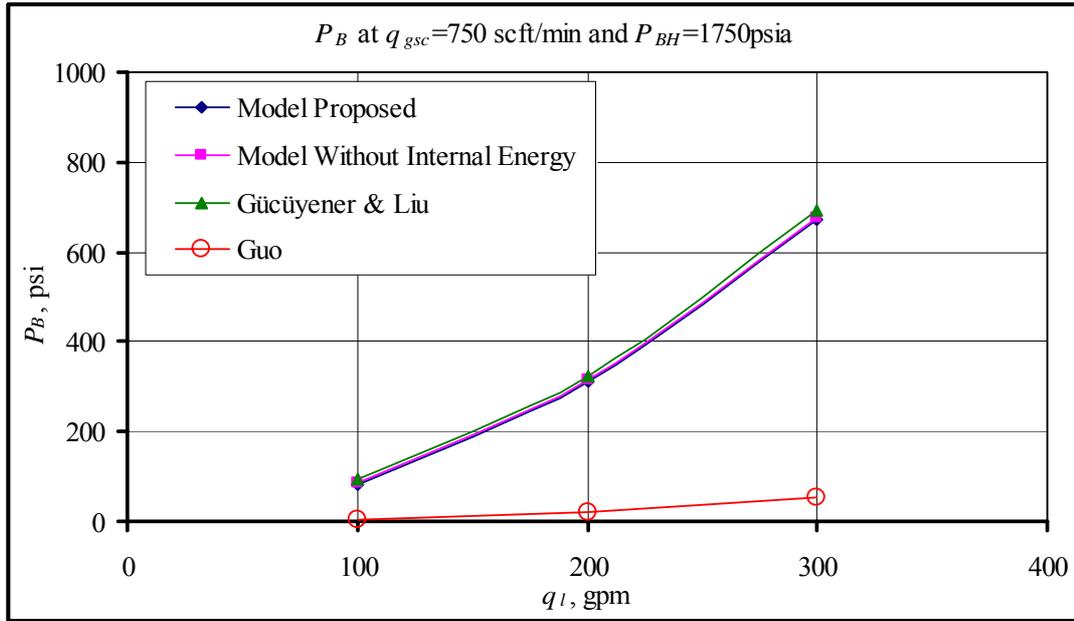


Figure 23  $P_B$  Values at  $q_{gsc}=750$  scft/min and  $P_{BH}=1750$ psia

## **CHAPTER 6**

### **CONCLUSIONS**

In this study, a mathematical model has been developed for calculating bit pressure drop across a nozzle for two phase drilling fluids. The general energy equation and the sound wave transmission concept are developing the basis of the proposed model. Developed model is valid for both sonic and subsonic flow. Air – water mixture, have been used for testing the performance of the proposed model. The obtained conclusions are listed below:

1. It is a general assumption that flow pattern is disperse bubbly through the bit nozzles. In this study, it has been proved that this assumption is valid.
2. The assumptions of all components are moving with the same velocity for a particular arbitrary point, and no slippage between the phases at the nozzle throat, are verified.
3. Since the sonic flow through the nozzle is not desired in drilling operations, the model calculates the pressure drop at the bit for subsonic conditions. If the flow

is sonic, the model changes the liquid and gas flow rates to achieve subsonic conditions.

4. A sensitivity analysis is conducted for investigating the effects of bottom hole pressure, liquid and gas flow rates, temperature, nozzle size, and bit size on pressure drop at the bit. The followings are concluded:
  - i. Bottom hole pressure has a significant effect on bit pressure drop across the nozzles. The bit pressure drop reduces with an increase in bottom hole pressure.
  - ii. Liquid rate has a noticeable effect on bit pressure drop. The liquid phase in the two-phase fluid mixture is incompressible. The amount of incompressible phase has a great effect on pressure drop through the nozzle.
  - iii. The influence of gas flow rate on pressure drop across the nozzle is not as significant as the liquid flow rate. However, still an increase in the pressure drop is observed as the gas flow rate increased.
  - iv. Bottom hole temperature has a very slight effect on bit pressure drop.
  - v. The nozzle size has a strong influence on bit pressure drop. As the nozzle size is decreased, a significant increase in bit pressure drop is observed.

- vi. The bit size has no effect on bit pressure drop. The hydraulic calculations for pipe and annular flow take the bit size into consideration. However, the hydraulic calculations for pipe and annular flow are out of scope of this study.
5. The proposed model can calculate the bit pressure drop with a reasonable accuracy. The developed model has a maximum 9% difference with the Gücüyener [17] and Liu [18] model results.
6. The superior part of this proposed model when compared with the other models is to be taken into the internal energy and temperature changes consideration. Taking into consideration of the internal energy improves the accuracy of the bit pressure drop calculations.
7. Most widely used model in the drilling industry, Gou [15] model under estimates the pressure drop at the bit and Gücüyener [17] and Liu [18] model overestimates the pressure drop at the bit.
  - i. Guo [15] model is valid only for high gas flow rates and high gas concentrations. The results of the Guo [15] model give lower results than the Gücüyener [17] and Liu [18] model and the proposed model. Thus, Therefore Gou [15] model under estimates the pressure drop at the bit.

- ii. Internal energy, compressibility factor, temperature changes and initial kinetic energy term do not have any effect on Gücüyener [17] and Liu [18] model results, because, those factors are not included in Gücüyener [17] and Liu [18] model. Therefore, absence of those factors result in the pressure drop of the bit, as seen in Gücüyener [17] and Liu [18] model results. Indeed, the elimination of initial kinetic energy term in their model is the major factor of the variation with the proposed model.
8. When internal energy term is ignored, the proposed model gives closer results to Gücüyener [17] and Liu [18] model. The internal energy changes in the proposed model reduce the bit pressure drop values about 3.3%. Thus, the proposed model is more sensitive.

## **CHAPTER 7**

### **RECOMMENDATIONS**

In this study, a mathematical model has been developed to calculate bit pressure drop across a nozzle for two phase drilling fluids. The general energy equation is used during the development of this model. Air – water mixtures have been used for testing the performance of the proposed model. Some recommendations are listed below.

- The proposed model has been compared with the existing model. However, to verify validity of the proposed model, the results of the model must be compared with experimental results.
- Some calculations have been conducted for air – water mixture. The results are used for verification of the model with existing models. Indeed, it would be better to verify the model for different liquid and gas component than air and water, i.e., different gases such as CO<sub>2</sub>, N<sub>2</sub>, and different liquids, such as non-Newtonian fluids.

- The proposed model should be improved for soluble fluids. In this study, it is assumed that, the gas phase of the fluid is not soluble in liquid phase of the fluid. In this study, the solubility effect is not taken into consideration.
- The liquid heat capacity at constant pressure and volume values should be correlated. The heat capacity of liquid is assumed to be constant. Yet, it shows small changes with temperature and pressure.

## CHAPTER 8

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## APPENDIX A

### CORRELATION OF AIR PROPERTIES

#### A.1. Correlation for Air Heat Capacity at Constant Pressure Values

The developed computer program based on the proposed model also calculates the heat capacity values measured by Çengel and Bole [21], and Hodgman [22]. The measured values are numerically processed by the computer and the results are obtained at given pressures and temperatures.

Table 2 Constant Pressure Heat Capacity Values

P, psi	Temperature			
	-58 °F	32 °F	122 °F	212°F
<b>14.7</b>	186.7139	186.5582	186.8687	187.5387
<b>147</b>	190.3492	189.6079	189.5775	190.2572
<b>294</b>	196.1764	194.3088	192.9859	192.2077
<b>588</b>	213.2961	202.7014	197.0065	196.2121
<b>1029</b>	242.8665	218.6682	204.4515	200.2164
<b>1470</b>	301.3886	250.2086	211.5842	202.3239
<b>3234</b>		714.9144	230.4158	221.0778

$C_p$ , ft-lbf/lbm-°F

## A.2. Gas Compressibility Factor For Air

In this study, the air compressibility factor,  $z$ , is estimated using Papay's [23] method, which is given by.

$$z = 1.0 - (P_r / T_r)[0.36748758 - 0.04188423(P_r / T_r)] \dots\dots\dots (A.1)$$

In this equation, the reduced pressure,  $P_r$ , and the reduced temperature,  $T_r$ , of air are given by,

$$P_r = P / 547 \dots\dots\dots (A.2)$$

and

$$T_r = (T + 460) / 239 \dots\dots\dots (A.3)$$

respectively.

## APPENDIX B

### EXISTING MODELS

#### B.1. Gücüyener and Liu Model

Gücüyener [17] and Liu [18] developed an equation for determination of pressure drop across the nozzle. In their studies, the general energy equation for two-phase flow through the bit nozzles was used for developing the bit pressure drop equation.

$$v_f dP + \frac{1}{g_c} v_f d v_f + dE + \frac{g}{g_c} dZ = 0 \dots\dots\dots (A.4)$$

In their work, it was assumed that the volumetric velocity of the fluid is almost zero at the upstream conditions. Therefore, the upstream velocity of the fluid was assigned to zero. Moreover, the elevation changes and internal energy term in the general energy equation were neglected because of their slight effect on the calculations. Thus, the integration of eq. (A.4) becomes,

$$\int_{P_{up}}^{P_{BH}} v_f dP + \frac{1}{g_c} \int_0^{v_{nozzle}} \vartheta_f d\vartheta_f = 0 \dots\dots\dots (A.5)$$

After integration, eq. (A.5) becomes

$$1.08 \times 10^{-4} \left( \frac{v_{nozzle}}{C_N} \right)^2 + \frac{f_g}{\rho_g} P_{BH} \left( \ln \frac{P_{BH}}{P_{BH} - P_B} \right) - \frac{f_l}{\rho_l} (P_B) = 0 \dots\dots\dots (A.6)$$

The results of the equation were compared with those of the proposed model. The results of the Gücüyener [17] and Liu [18] equation overestimate the pressure drop of the bit.

## B.2. Bit Pressure Drop Determination Equation By Guo

Guo, et. al. [15] developed an equation for estimating the pressure drop of a multiphase fluid flow through a nozzle. From the first law of thermodynamics, the following equation was obtained.

$$\frac{dP}{\rho_f} + \frac{v_f}{g_c} dv_f + \frac{g}{g_c} dZ = 0 \dots\dots\dots (A.7)$$

According to their study, the elevation changes and energy losses caused by friction were neglected. So, eq. (A.7) became

$$dP = -\frac{\rho_f}{g_c} v_f dv_f \dots\dots\dots (A.8)$$

Integration of eq. (A.8) yielded

$$P_{upstream} - P_{downstream} = \frac{\rho_m}{2g_c} (v_{f-downstream}^2 - v_{f-upstream}^2) \dots\dots\dots (A.9)$$

The velocity of a fluid can be determined as

$$v_f = \frac{\dot{m}_f}{\rho_f \times A_{nozzle}} \dots\dots\dots (A.10)$$

Thus, substituting eq. (A.10) into eq. (A.9) yields

$$P_B = \left( \frac{\dot{m}_f}{A_{nozzle}} \right)^2 \frac{1}{g_c \times 144} (v_{f-downstream} - v_{f-upstream}) \dots\dots\dots (A.11)$$

Equation (A.11) depends on the specific volume of the fluid. For incompressible mixtures or fluids, the specific volume difference in the equation reduced to zero. Therefore, the equation is valid only for compressible fluids.

## APPENDIX C

### ITERATION ERROR ANALYSIS

The developed created computer program based on the proposed model is used to calculate the pressure drop through a nozzle. The calculation process contains iterations. After each iteration, the program calculates the difference between the consecutive iterated values and compares the difference with desired error. The desired iteration error value is defined by the user, and assigned to get more accurate results. The iteration error values and obtained result values are listed in Table 3. For higher error values, the computer program gives results in a long period of time, although there is not a significant difference between the results. Therefore, for calculation of the pressure drop values, the error value of  $1.0 \times 10^{-6}$  is selected for iteration error.

Table 3 Iteration Error Analysis

<b>Error</b>	$1.0 \times 10^{-4}$	$1.0 \times 10^{-5}$	$1.0 \times 10^{-6}$	$1.0 \times 10^{-7}$
<b>Calculated <math>P_B</math></b>	1209.0864745	1241.6502137	1245.0172463	1245.3350807

## APPENDIX D

### VARIATION ANALYSIS

The following statistical parameters have been used to compute the accuracy of the investigated correlation.

- Average absolute percent relative deviation:

$$APD = (1/k) \sum_{i=1}^k PD_i \dots\dots\dots (A.12)$$

where

$$PD_i = ((P_{Bo} - P_{Bm}) / P_{Bo}) \times 100 \dots\dots\dots (A.13)$$

$$k = 1, 2, 3, 4, \dots$$

$P_{Bm}$  is bit pressure calculated by the proposed model

$P_{Bo}$  is bit pressure calculated by the existing model

- Standard deviation:

$$SD = \sqrt{\frac{k \left( \left( \sum_{i=1}^k PD_i^2 \right) - \left( \sum_{i=1}^k PD_i \right)^2 \right)}{k(k-1)}} \dots\dots\dots (A.14)$$

## APPENDIX E

### CALCULATED PRESSURE DROP VALUES AND VARIATION ANALYSIS

The following values, listed in the tables and plotted on the figures, are used to compare the proposed model with the existing models.

Table 4  $P_B$  values of Case 1 at  $q_l=100$  gal/min and  $P_{BH}=250$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	66.310	66.310	67.350	0.000
200	140.410	140.650	143.150	47.040
377	222.250	223.783	226.380	117.680

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.544	1.544		
200	0.171	0.171	1.914	1.914	-198.491	198.491
377	0.685	0.685	1.824	1.824	-88.860	88.860
<b>Average</b>	<b>0.285</b>	<b>0.285</b>	<b>1.761</b>	<b>1.761</b>	<b>-143.675</b>	<b>143.675</b>
<b>SD</b>	<b>0.357</b>	<b>0.357</b>	<b>0.193</b>	<b>0.193</b>	<b>77.521</b>	<b>77.521</b>

Table 5  $P_B$  values of Case 1 at  $q_l=100$  gal/min and  $P_{BH}=500$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	66.310	66.310	67.350	0.000
200	96.200	96.850	99.230	10.580
500	147.580	148.360	152.550	38.210
750	196.410	196.710	202.590	72.010
1000	249.720	250.400	258.400	115.490
1397	344.540	350.786	360.650	203.980

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.544	1.544		
200	0.671	0.671	3.054	3.054	-809.263	809.263
500	0.526	0.526	3.258	3.258	-286.234	286.234
750	0.153	0.153	3.050	3.050	-172.754	172.754
1000	0.272	0.272	3.359	3.359	-116.227	116.227
1397	1.781	1.781	4.467	4.467	-68.909	68.909
<b>Average</b>	<b>0.567</b>	<b>0.567</b>	<b>3.122</b>	<b>3.122</b>	<b>-290.677</b>	<b>290.677</b>
<b>SD</b>	<b>0.643</b>	<b>0.643</b>	<b>0.936</b>	<b>0.936</b>	<b>301.028</b>	<b>301.028</b>

Table 6  $P_B$  values of Case 1 at  $q_l=100$  gal/min and  $P_{BH}=750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without InternalEnergy	Güçüyener & Liu	Guo
0	66.310	66.310	67.350	0.000
200	83.420	84.390	86.910	4.380
500	112.530	114.150	118.410	14.700
750	139.540	141.190	146.750	26.940
1000	168.930	170.240	177.130	42.660
1500	234.200	234.600	244.520	85.100

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.544	1.544		
200	1.149	1.149	4.016	4.016	-1804.566	1804.566
500	1.419	1.419	4.966	4.966	-665.510	665.510
750	1.169	1.169	4.913	4.913	-417.966	417.966
1000	0.770	0.770	4.629	4.629	-295.992	295.992
1500	0.171	0.171	4.221	4.221	-175.206	175.206
<b>Average</b>	<b>0.780</b>	<b>0.780</b>	<b>4.048</b>	<b>4.048</b>	<b>-671.848</b>	<b>671.848</b>
<b>SD</b>	<b>0.579</b>	<b>0.579</b>	<b>1.283</b>	<b>1.283</b>	<b>658.686</b>	<b>658.686</b>

Table 7  $P_B$  values of Case 1 at  $q_l=100$  gal/min and  $P_{BH}=750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	66.300	66.300	67.300	0.000
200	73.600	75.000	77.800	1.400
500	86.500	89.300	94.200	4.400
750	99.000	102.400	108.600	7.700
1000	112.600	116.300	123.800	11.700
1500	143.000	146.500	156.200	22.400

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.486	1.486		
200	1.867	1.867	5.398	5.398	-5157.143	5157.143
500	3.135	3.135	8.174	8.174	-1865.909	1865.909
750	3.320	3.320	8.840	8.840	-1185.714	1185.714
1000	3.181	3.181	9.047	9.047	-862.393	862.393
1500	2.389	2.389	8.451	8.451	-538.393	538.393
<b>Average</b>	<b>2.315</b>	<b>2.315</b>	<b>6.899</b>	<b>6.899</b>	<b>-1921.910</b>	<b>1921.910</b>
<b>SD</b>	<b>1.265</b>	<b>1.265</b>	<b>2.965</b>	<b>2.965</b>	<b>1874.070</b>	<b>1874.070</b>

Table 8  $P_B$  values of Case 1 at  $q_l=100$  gal/min and  $P_{BH}=1750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	66.300	66.300	67.300	0.000
200	69.900	71.300	74.100	0.700
500	76.500	79.600	84.700	2.000
750	83.100	87.200	93.900	3.300
1000	90.800	95.500	103.500	5.000
1500	108.400	113.500	123.700	9.100

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.486	1.486		
200	1.964	1.964	5.668	5.668	-9885.714	9885.714
500	3.894	3.894	9.681	9.681	-3725.000	3725.000
750	4.702	4.702	11.502	11.502	-2418.182	2418.182
1000	4.921	4.921	12.271	12.271	-1716.000	1716.000
1500	4.493	4.493	12.369	12.369	-1091.209	1091.209
<b>Average</b>	<b>3.329</b>	<b>3.329</b>	<b>8.829</b>	<b>8.829</b>	<b>-3767.221</b>	<b>3767.221</b>
<b>SD</b>	<b>1.952</b>	<b>1.952</b>	<b>4.385</b>	<b>4.385</b>	<b>3557.604</b>	<b>3557.604</b>

Table 9  $P_B$  values of Case 1 at  $q_l=200$  gal/min and  $P_{BH}=500$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0.0	265.260	265.260	269.380	0.000
200	334.840	335.400	341.340	51.590
500	452.230	453.300	462.510	154.790
750	563.000	564.200	576.900	261.580
757.38	568.900	568.958	580.500	264.990

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0.0	0.000	0.000	1.529	1.529		
200	0.167	0.167	1.904	1.904	-549.041	549.041
500	0.236	0.236	2.223	2.223	-192.157	192.157
750	0.213	0.213	2.409	2.409	-115.231	115.231
757.38	0.010	0.010	1.998	1.998	-114.687	114.687
<b>Average</b>	<b>0.125</b>	<b>0.125</b>	<b>2.013</b>	<b>2.013</b>	<b>-242.779</b>	<b>242.779</b>
<b>SD</b>	<b>0.112</b>	<b>0.112</b>	<b>0.334</b>	<b>0.334</b>	<b>207.392</b>	<b>207.392</b>

Table 10  $P_B$  values of Case 1 at  $q_l=200$  gal/min and  $P_{BH}=750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	265.300	265.300	269.400	0.000
200	305.600	306.300	312.000	24.580
500	370.600	372.000	380.400	71.230
750	429.000	430.700	441.500	118.770
1000	491.400	493.200	506.800	173.920
1500	628.500	630.100	650.900	305.900

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.522	1.522		
200	0.229	0.229	2.051	2.051	-1143.287	1143.287
500	0.376	0.376	2.576	2.576	-420.286	420.286
750	0.395	0.395	2.831	2.831	-261.202	261.202
1000	0.365	0.365	3.039	3.039	-182.544	182.544
1500	0.254	0.254	3.441	3.441	-105.459	105.459
<b>Average</b>	<b>0.270</b>	<b>0.270</b>	<b>2.577</b>	<b>2.577</b>	<b>-422.556</b>	<b>422.556</b>
<b>SD</b>	<b>0.149</b>	<b>0.149</b>	<b>0.695</b>	<b>0.695</b>	<b>419.419</b>	<b>419.419</b>

Table 11  $P_B$  values of Case 1 at  $q_l=200$  gal/min and  $P_{BH}=1250$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	265.300	265.300	269.400	0.000
200	284.800	285.700	291.300	9.200
500	315.700	317.600	325.400	25.400
750	343.000	345.400	355.000	41.100
1000	371.500	374.200	385.600	59.000
1500	431.800	434.800	450.100	101.100

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.522	1.522		
200	0.315	0.315	2.231	2.231	-2995.652	2995.652
500	0.598	0.598	2.981	2.981	-1142.913	1142.913
750	0.695	0.695	3.380	3.380	-734.550	734.550
1000	0.722	0.722	3.657	3.657	-529.661	529.661
1500	0.690	0.690	4.066	4.066	-327.102	327.102
<b>Average</b>	<b>0.503</b>	<b>0.503</b>	<b>2.973</b>	<b>2.973</b>	<b>-1145.976</b>	<b>1145.976</b>
<b>SD</b>	<b>0.289</b>	<b>0.289</b>	<b>0.947</b>	<b>0.947</b>	<b>1077.147</b>	<b>1077.147</b>

Table 12  $P_B$  values of Case 1 at  $q_l=200$  gal/min and  $P_{BH}=1750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	265.300	265.300	269.400	0.000
200	277.000	277.900	283.400	4.600
500	295.600	297.500	304.900	12.400
750	311.800	314.400	323.300	19.800
1000	328.700	331.800	342.300	27.900
1500	364.300	368.000	381.600	46.600

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.522	1.522		
200	0.324	0.324	2.258	2.258	-5921.739	5921.739
500	0.639	0.639	3.050	3.050	-2283.871	2283.871
750	0.827	0.827	3.557	3.557	-1474.747	1474.747
1000	0.934	0.934	3.973	3.973	-1078.136	1078.136
1500	1.005	1.005	4.534	4.534	-681.760	681.760
<b>Average</b>	<b>0.622</b>	<b>0.622</b>	<b>3.149</b>	<b>3.149</b>	<b>-2288.051</b>	<b>2288.051</b>
<b>SD</b>	<b>0.391</b>	<b>0.391</b>	<b>1.115</b>	<b>1.115</b>	<b>2115.965</b>	<b>2115.965</b>

Table 13  $P_B$  values of Case 1 at  $q_l=300$  gal/min and  $P_{BH}=500$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0.0	596.800	596.800	606.100	0.000
200	714.300	714.900	726.400	112.910
500	910.900	912.500	928.100	312.580
512.0	921.440	921.970	936.800	321.230

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0.0	0.000	0.000	1.534	1.534		
200	0.084	0.084	1.666	1.666	-532.628	532.628
500	0.175	0.175	1.853	1.853	-191.413	191.413
512.0	0.057	0.057	1.640	1.640	-186.847	186.847
<b>Average</b>	<b>0.079</b>	<b>0.079</b>	<b>1.673</b>	<b>1.673</b>	<b>-303.630</b>	<b>303.630</b>
<b>SD</b>	<b>0.073</b>	<b>0.073</b>	<b>0.133</b>	<b>0.133</b>	<b>198.331</b>	<b>198.331</b>

Table 14  $P_B$  values of Case 1 at  $q_l=300$  gal/min and  $P_{BH}=750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	596.800	596.800	606.100	0.000
200	664.400	665.000	676.400	59.260
500	772.100	773.600	788.800	161.350
750	868.200	870.300	889.200	257.700
1000	970.100	972.800	996.000	363.400
1142.1	1036.440	1036.570	1059.800	427.400

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.534	1.534		
200	0.090	0.090	1.774	1.774	-1021.161	1021.161
500	0.194	0.194	2.117	2.117	-378.525	378.525
750	0.241	0.241	2.362	2.362	-236.903	236.903
1000	0.278	0.278	2.600	2.600	-166.951	166.951
1142.1	0.013	0.013	2.204	2.204	-142.499	142.499
<b>Average</b>	<b>0.136</b>	<b>0.136</b>	<b>2.099</b>	<b>2.099</b>	<b>-389.208</b>	<b>389.208</b>
<b>SD</b>	<b>0.119</b>	<b>0.119</b>	<b>0.389</b>	<b>0.389</b>	<b>365.013</b>	<b>365.013</b>

Table 15  $P_B$  values of Case 1 at  $q_l=300$  gal/min and  $P_{BH}=1250$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	596.800	596.800	606.100	0.000
200	629.600	630.300	641.400	24.500
500	680.400	682.000	696.000	64.900
750	724.400	726.600	743.100	102.100
1000	769.800	772.500	791.800	142.400
1500	865.000	868.500	894.000	232.000

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.534	1.534		
200	0.111	0.111	1.840	1.840	-2469.796	2469.796
500	0.235	0.235	2.241	2.241	-948.382	948.382
750	0.303	0.303	2.516	2.516	-609.500	609.500
1000	0.350	0.350	2.778	2.778	-440.590	440.590
1500	0.403	0.403	3.244	3.244	-272.845	272.845
<b>Average</b>	<b>0.233</b>	<b>0.233</b>	<b>2.359</b>	<b>2.359</b>	<b>-948.223</b>	<b>948.223</b>
<b>SD</b>	<b>0.153</b>	<b>0.153</b>	<b>0.624</b>	<b>0.624</b>	<b>886.530</b>	<b>886.530</b>

Table 16  $P_B$  values of Case 1 at  $q_l=300$  gal/min and  $P_{BH}=1750$ psia

$Q_{GSC}$ scft/min	Proposed Model	Model without Internal Energy	Güçüyener & Liu	Guo
0	596.800	596.800	606.100	0.000
200	616.800	617.400	628.200	12.900
500	647.200	648.800	662.000	33.800
750	673.500	675.700	690.900	52.600
1000	700.400	703.100	720.400	72.600
1500	756.100	759.600	781.300	116.500

$Q_{GSC}$ scft/min	Model without Internal Energy.		Güçüyener & Liu		Guo	
	PD	PD	PD	PD	PD	PD
0	0.000	0.000	1.534	1.534		
200	0.097	0.097	1.815	1.815	-4681.395	4681.395
500	0.247	0.247	2.236	2.236	-1814.793	1814.793
750	0.326	0.326	2.518	2.518	-1180.418	1180.418
1000	0.384	0.384	2.776	2.776	-864.738	864.738
1500	0.461	0.461	3.225	3.225	-549.013	549.013
<b>Average</b>	<b>0.252</b>	<b>0.252</b>	<b>2.351</b>	<b>2.351</b>	<b>-1818.072</b>	<b>1818.072</b>
<b>SD</b>	<b>0.175</b>	<b>0.175</b>	<b>0.623</b>	<b>0.623</b>	<b>1667.677</b>	<b>1667.677</b>

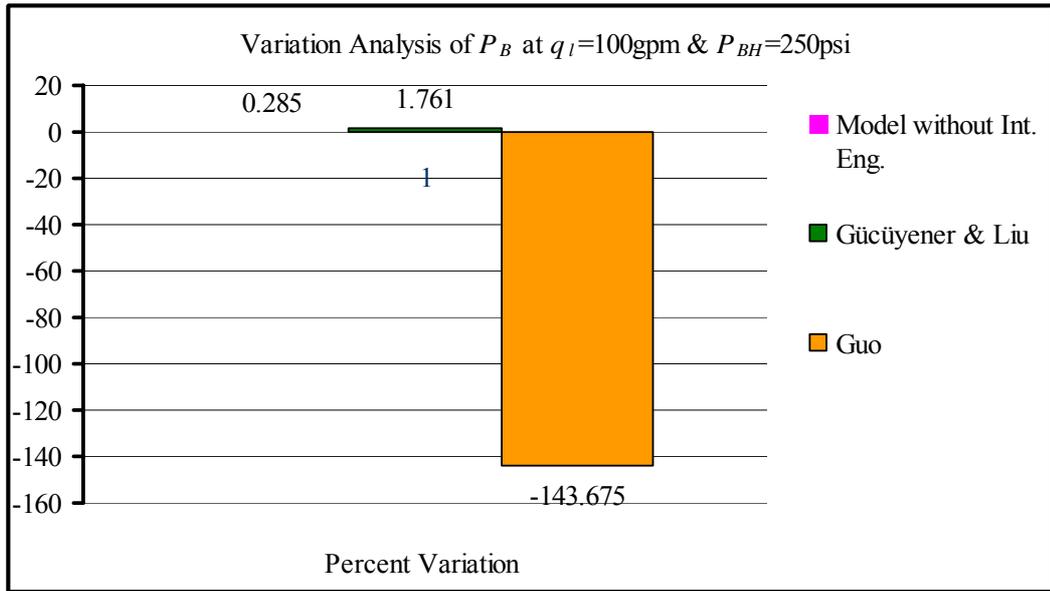


Figure 24 Variation Analysis of  $P_{BH}$  at  $q_l=100$  gal/min and  $P_{BH}=250$  psi

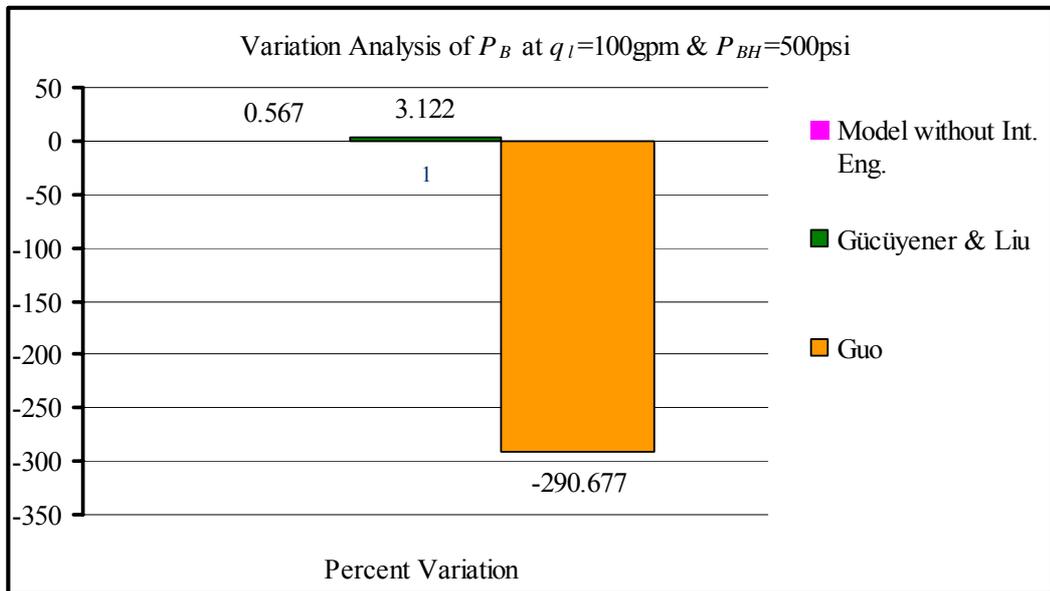


Figure 25 Variation Analysis of  $P_{BH}$  at  $q_l=100$  gal/min and  $P_{BH}=500$  psi

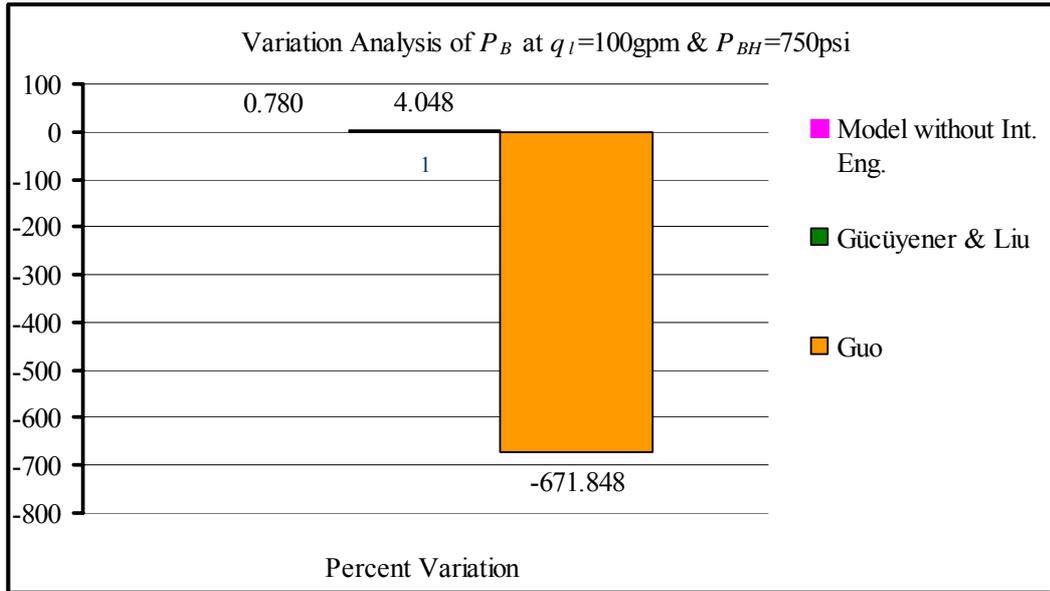


Figure 26 Variation Analysis of  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=750$  psi

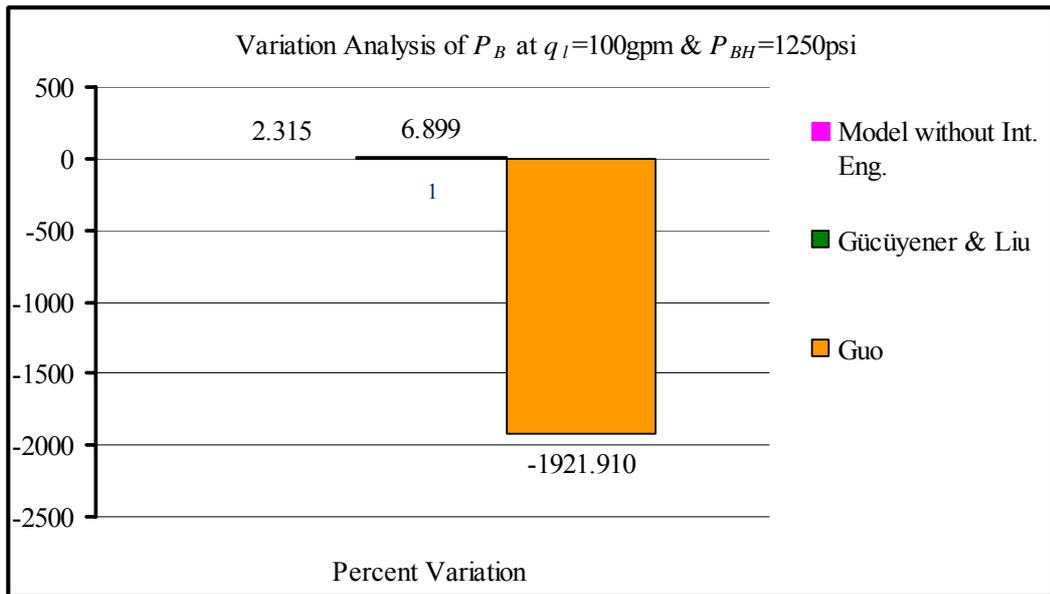


Figure 27 Variation Analysis of  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=1250$  psi

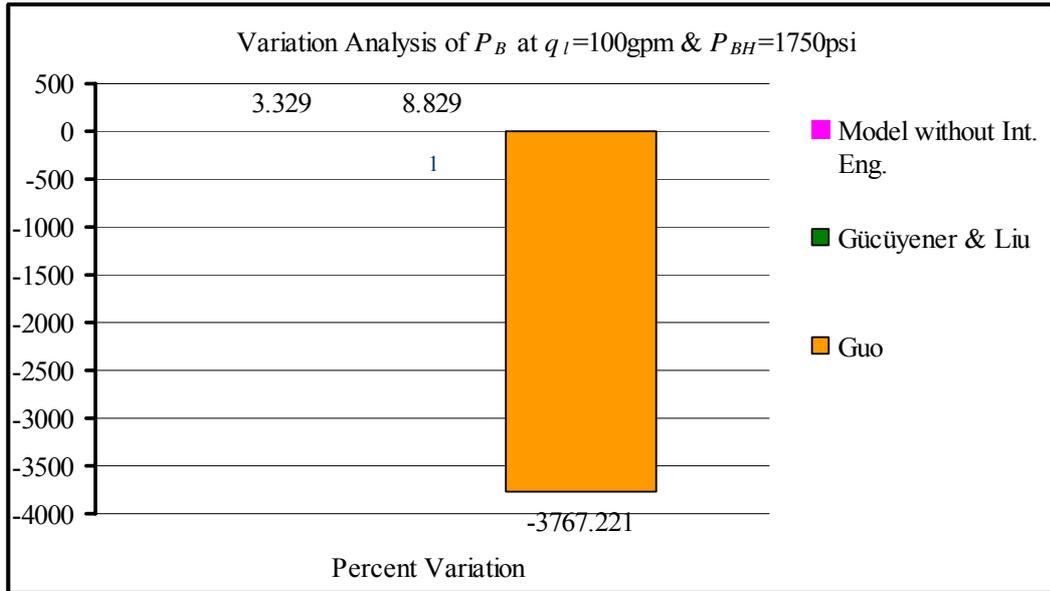


Figure 28 Variation Analysis of  $P_B$  at  $q_l=100$  gal/min and  $P_{BH}=1750$  psi

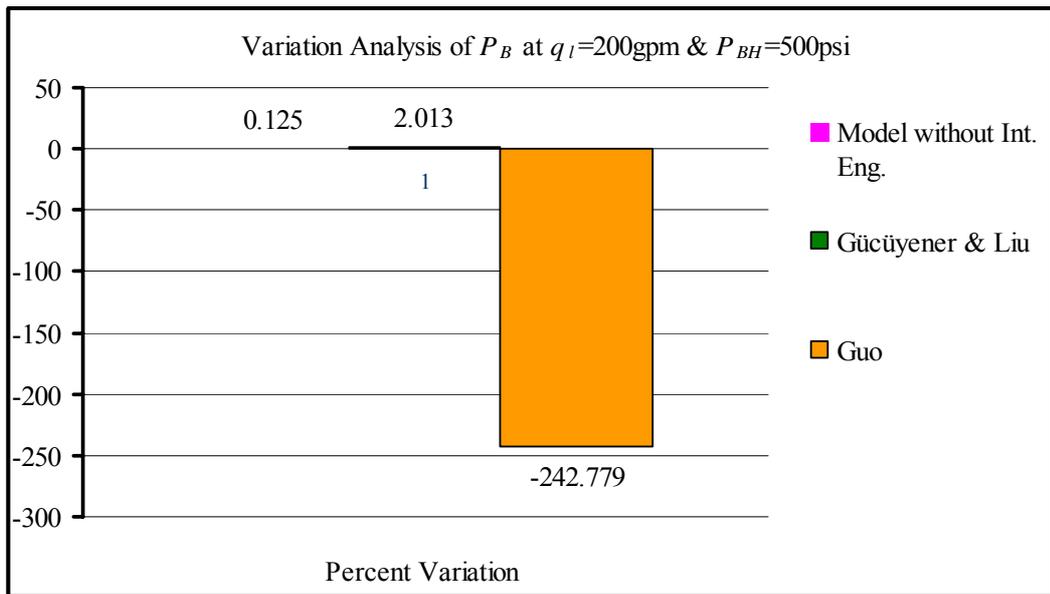


Figure 29 Variation Analysis of  $P_B$  at  $q_l=200$  gal/min and  $P_{BH}=500$  psi

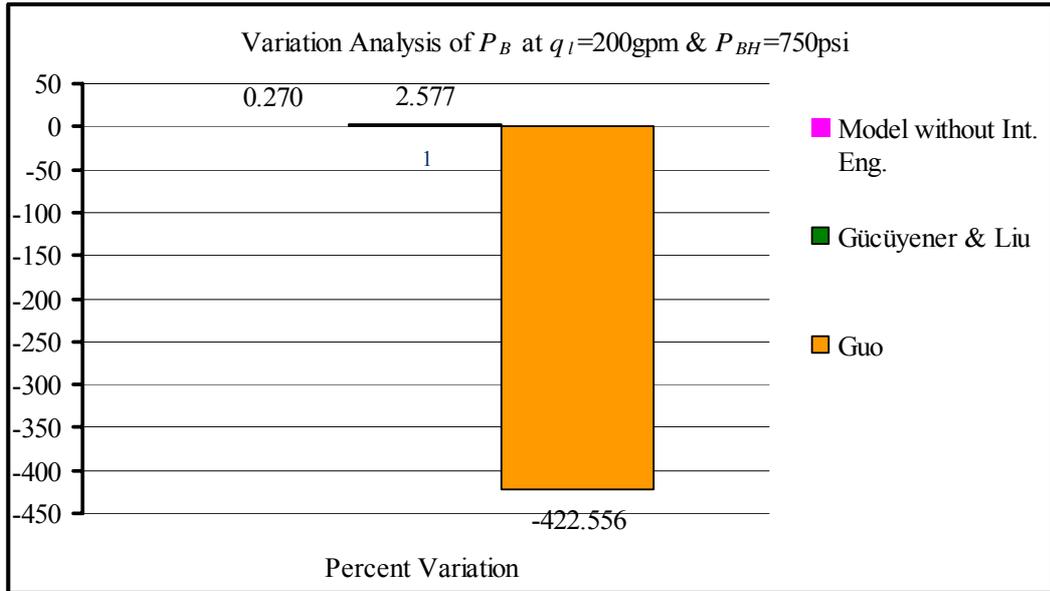


Figure 30 Variation Analysis of  $P_B$  at  $q_l=200$  gal/min and  $P_{BH}=750$  psi

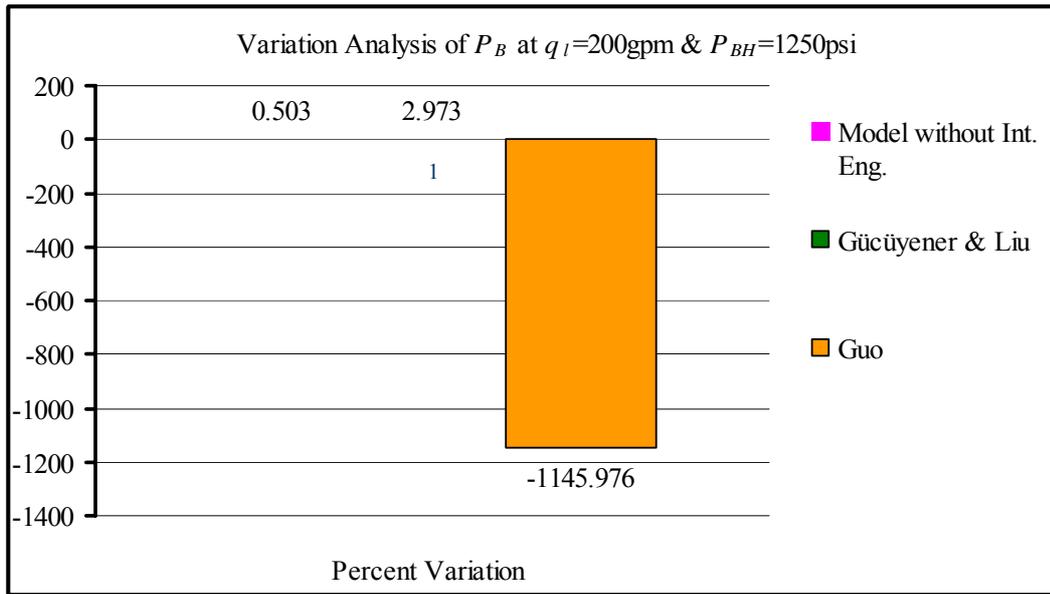


Figure 31 Variation Analysis of  $P_B$  at  $q_l=200$  gal/min and  $P_{BH}=1250$  psi

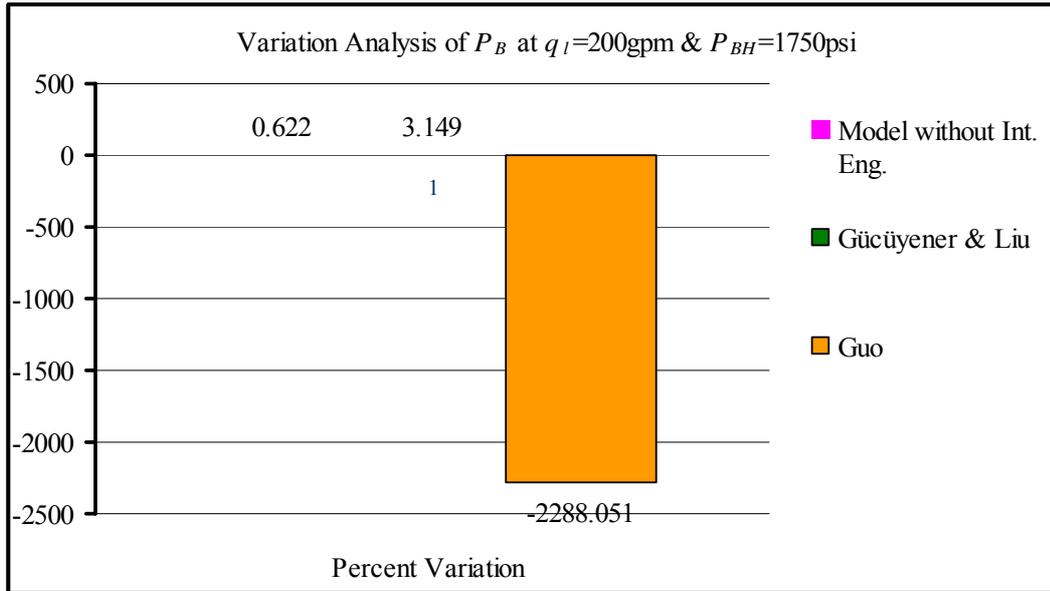


Figure 32 Variation Analysis of  $P_B$  at  $q_l=200$  gal/min and  $P_{BH}=1750$  psi

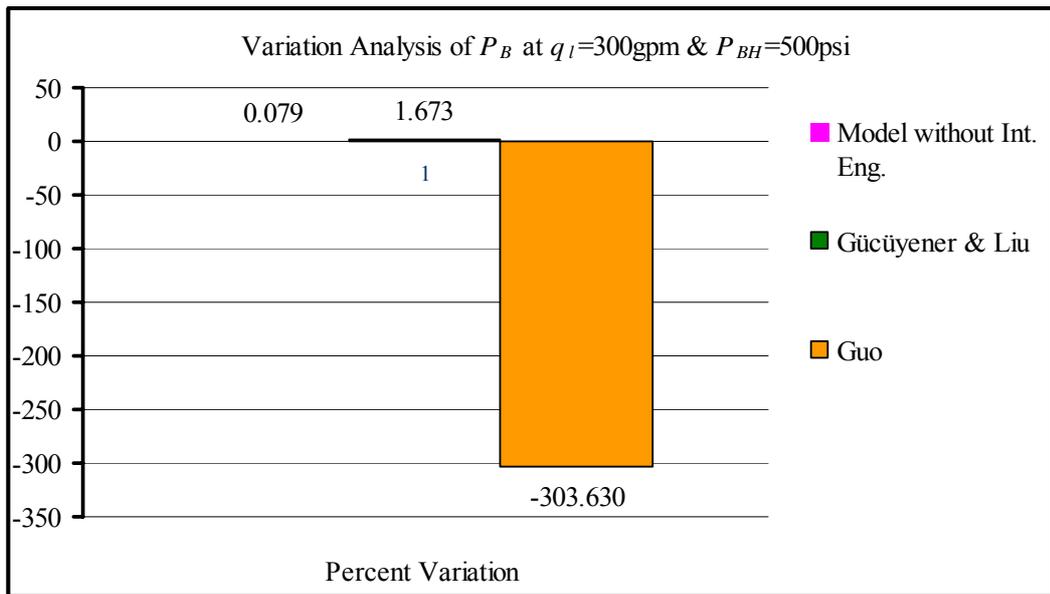


Figure 33 Variation Analysis of  $P_B$  at  $q_l=300$  gal/min and  $P_{BH}=500$  psi

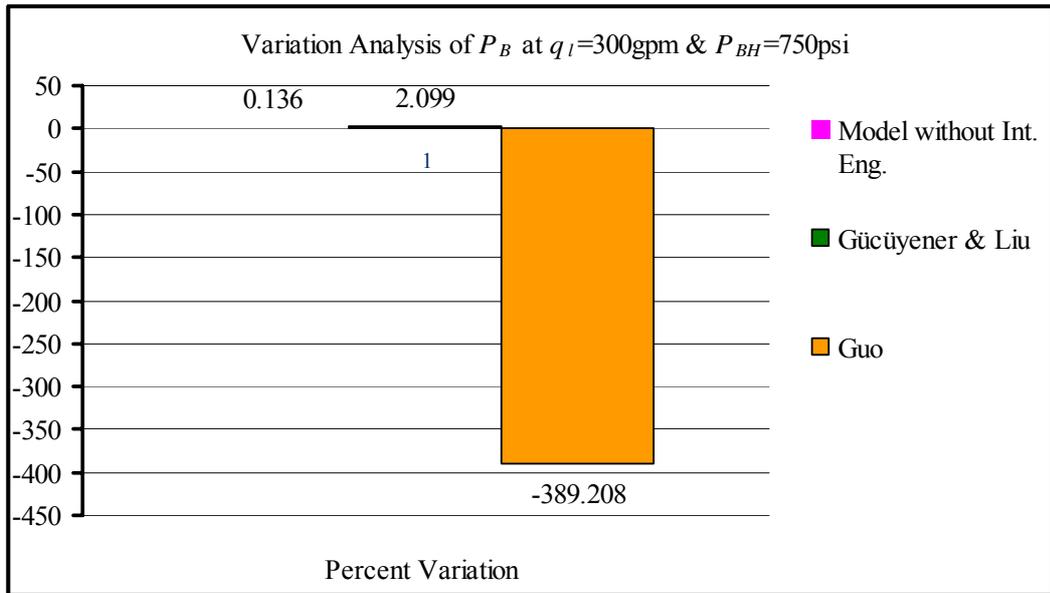


Figure 34 Variation Analysis of  $P_B$  at  $q_l=300$  gal/min and  $P_{BH}=750$  psi

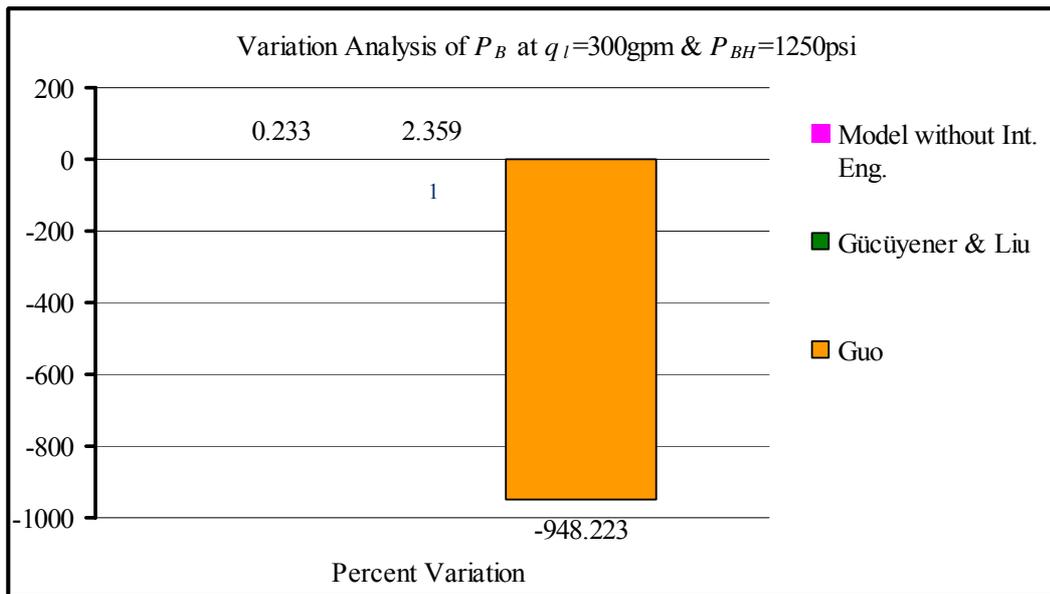


Figure 35 Variation Analysis of  $P_B$  at  $q_l=300$  gal/min and  $P_{BH}=1250$  psi

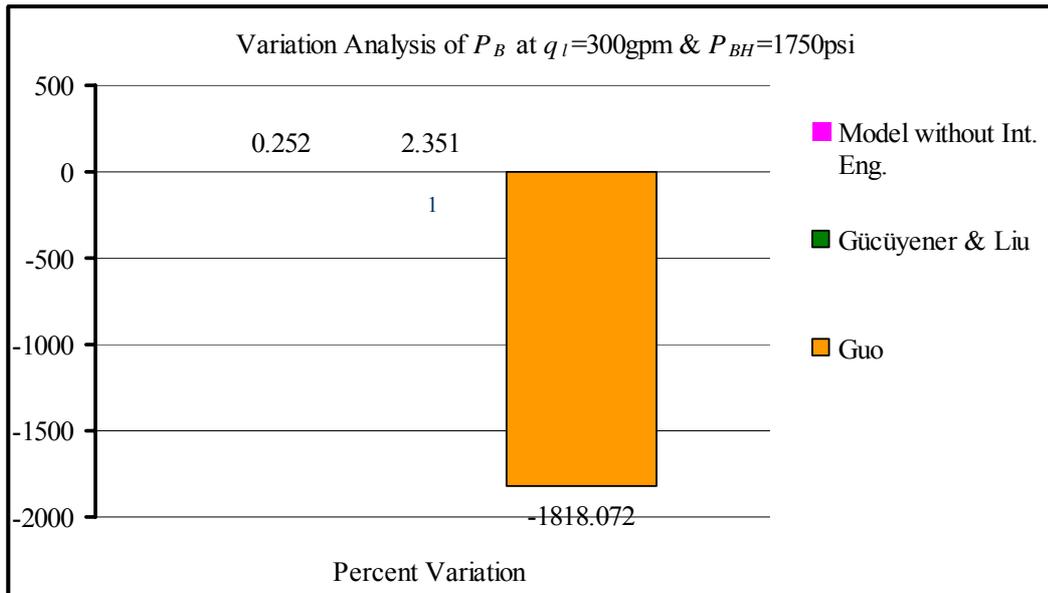


Figure 36 Variation Analysis of  $P_B$  at  $q_l=300$  gal/min and  $P_{BH}=1750$  psi